



# JAXA Agency Report

Keiji Imaoka

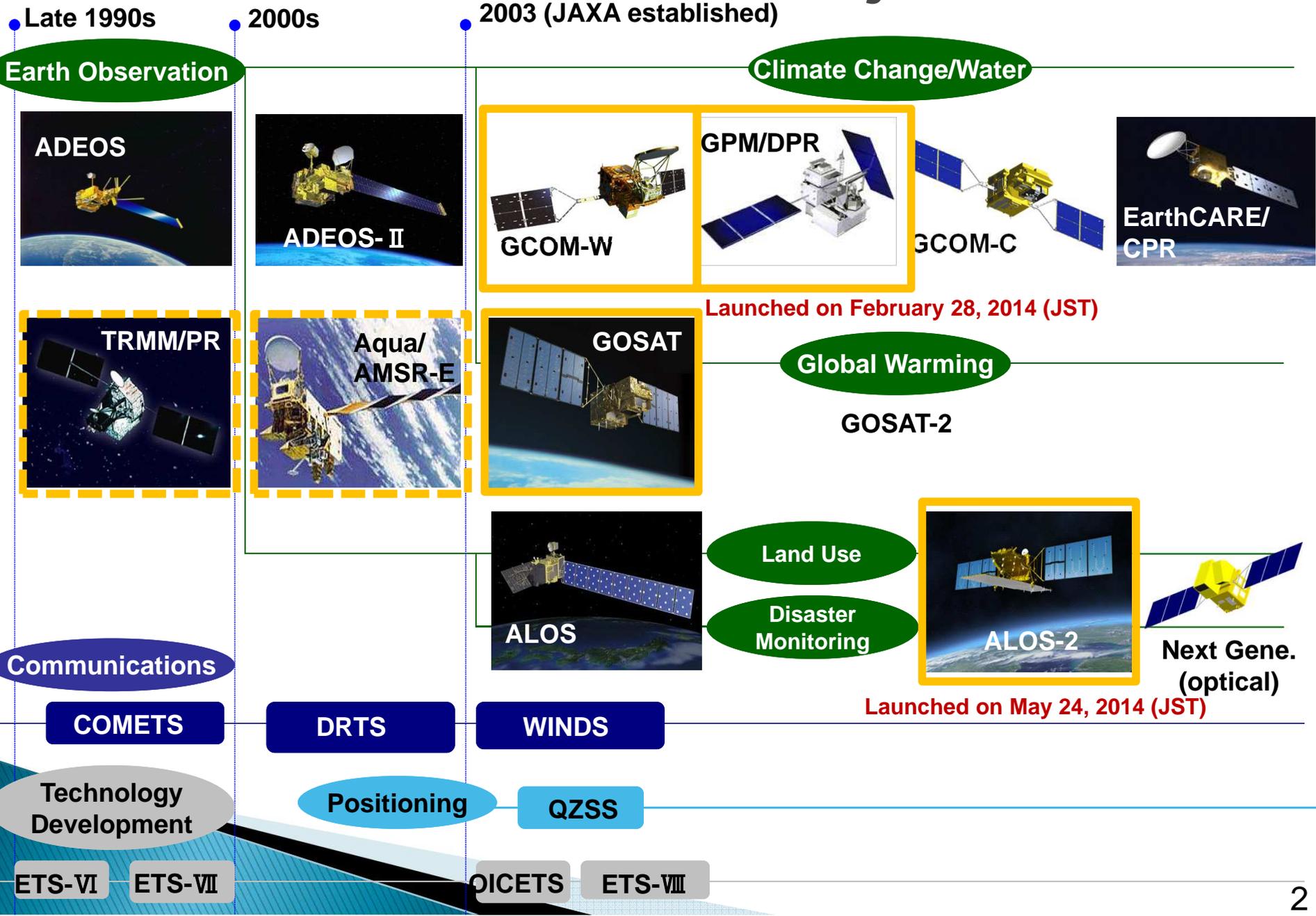
Earth Observation Research Center (EORC)  
Japan Aerospace Exploration Agency (JAXA)

GSICS/GRWG Meeting

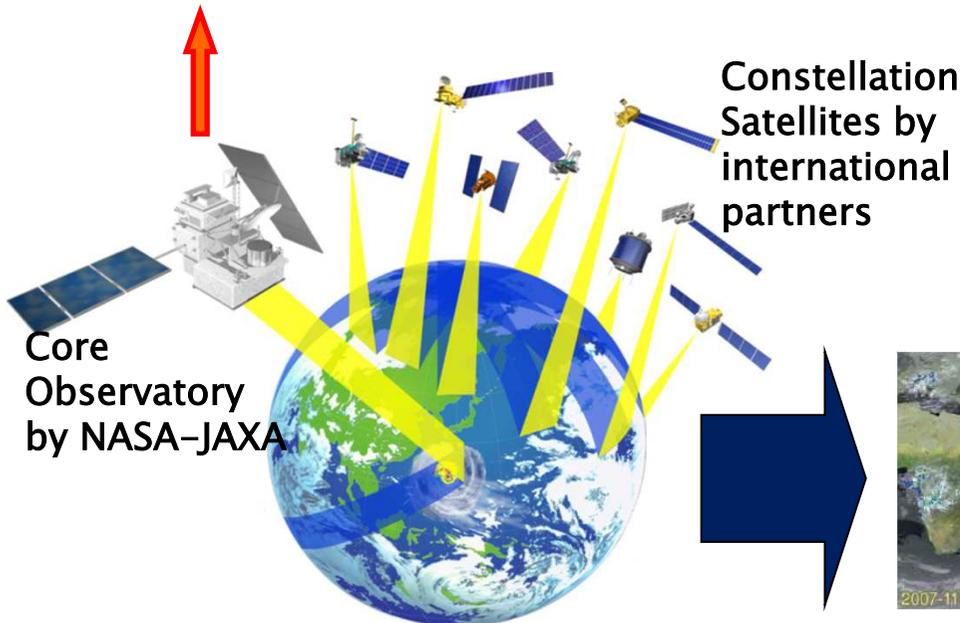
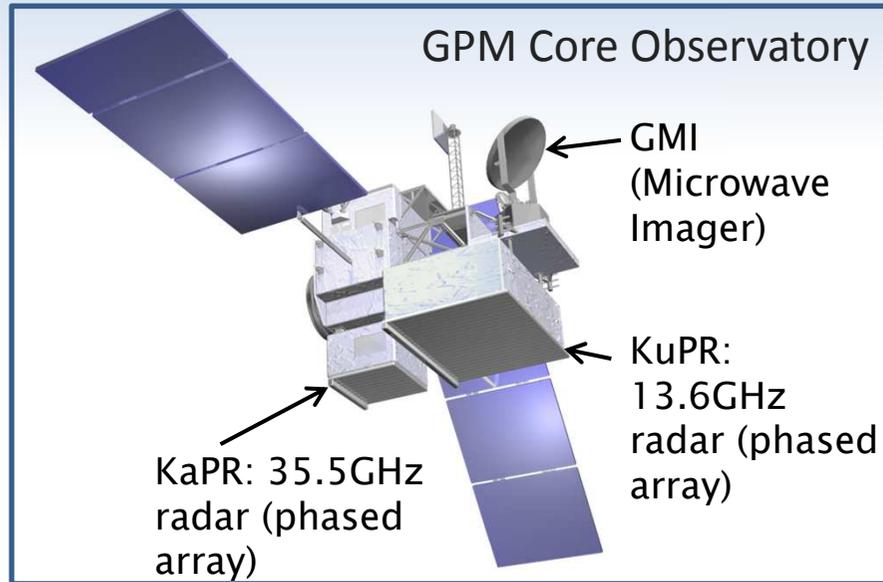
Delhi, India (remote participation)

March 17, 2015

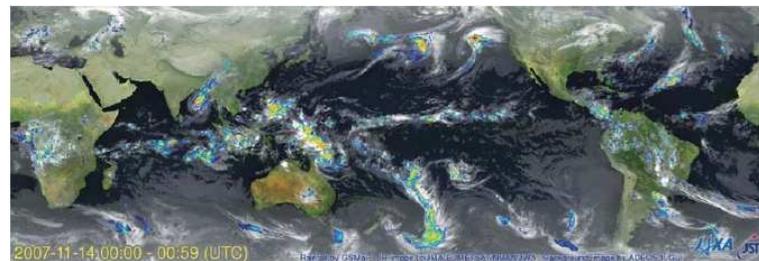
# JAXA Satellite Projects



# GPM



- GPM is an international mission consisting of the GPM Core Observatory and Constellation Satellites for high accurate and frequent global precipitation observation.
  - Core Observatory: developed under NASA and JAXA equal partnership.
  - Constellation satellites: provided by international partners (includes GCOM-W1).
- **Dual-frequency Precipitation Radar (DPR)**
  - developed by JAXA and NICT
  - DPR is composed of two radars: KuPR & KaPR
- GPM Core Observatory was successfully launched **on 28 Feb. 2014 (JST)**.



# Concept of DPR Calibration

The calibration of DPR is divided by ‘Internal calibration’ and ‘External calibration’  
Radio Frequency is not transmitted to out of DPR

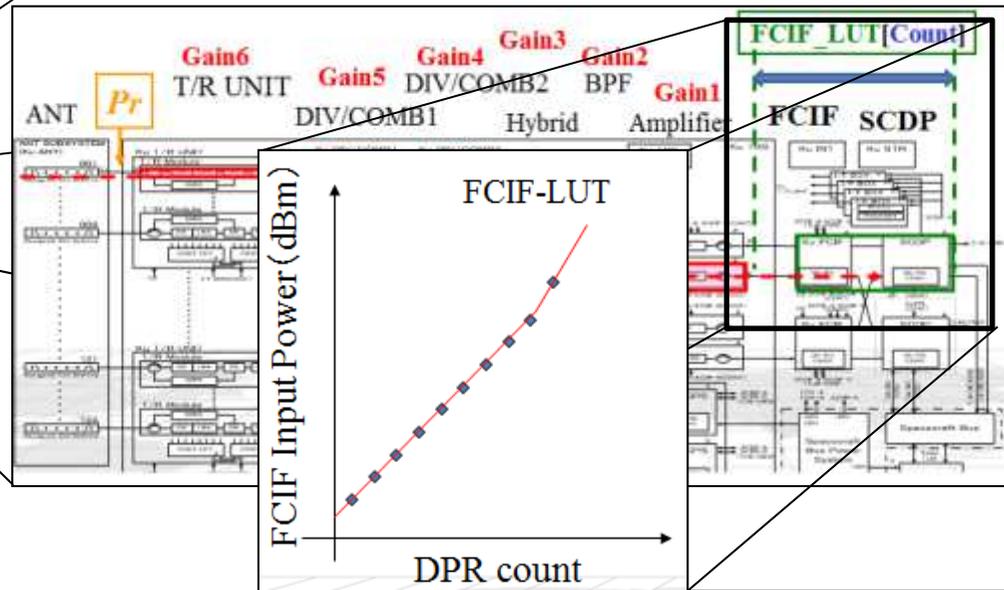
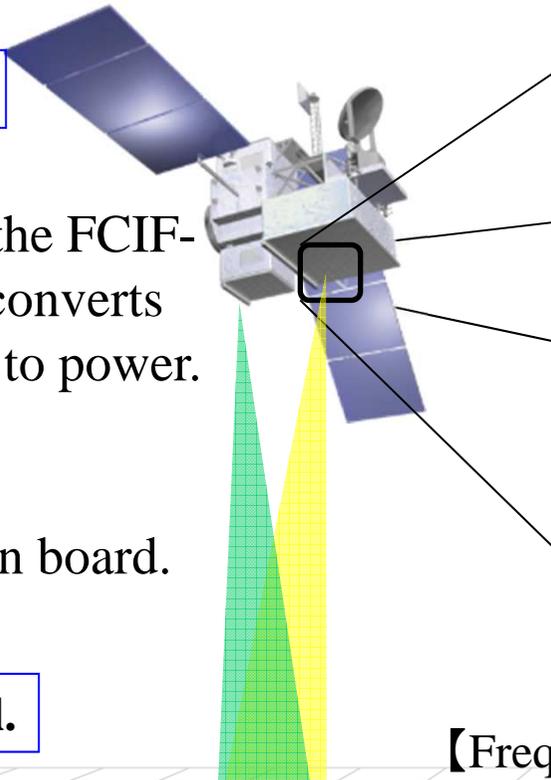
## Internal Cal.

### 【Purpose】

To calibrate the FCIF-LUT which converts digital count to power.

### 【Frequency】

Per a week on board.



## External Cal.

### 【Purpose】

To calibrate the absolute value of the DPR transmitted / received powers.

### 【Frequency】

External calibration will be planned 10 times in a year. During check out phase, the calibration have been conducted 10 times a month at JAXA tsukuba space center in Japan.

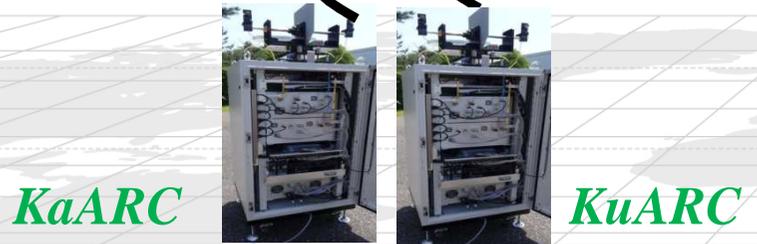
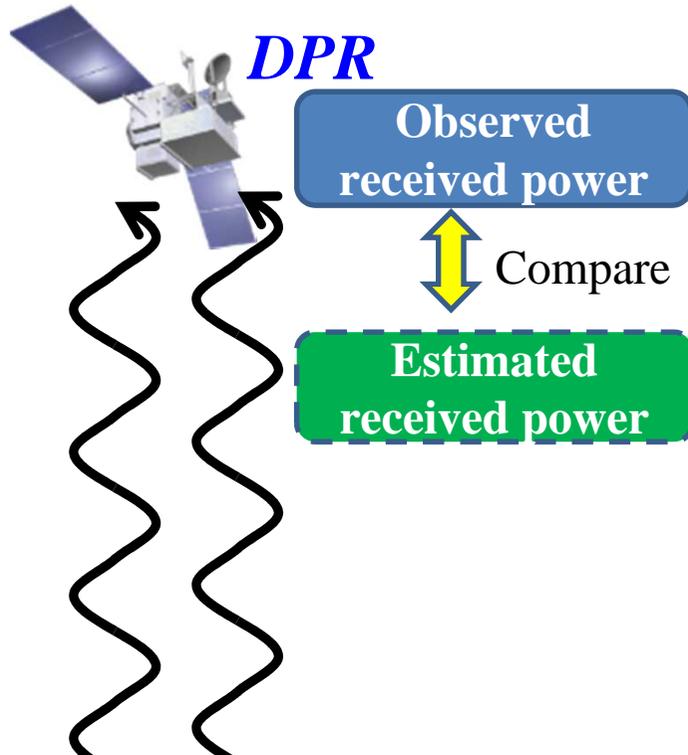


FCIF: Frequency Converter & IF

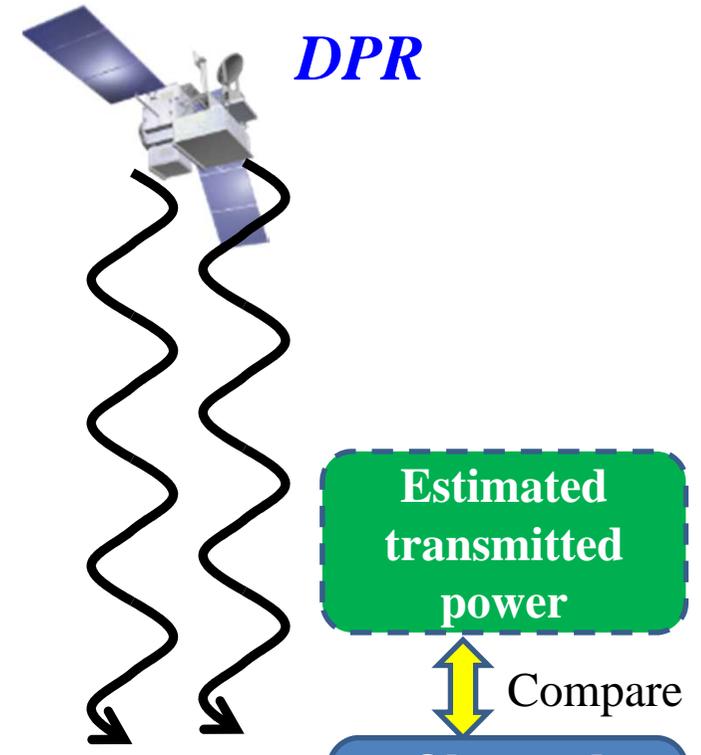
# Concept of DPR External Calibration



## DPR receiver system



## DPR transmitter system



To calibrate the received power and the transmitted power of DPR, the difference between observed power and estimated power is evaluated.

# Active Radar Calibrator

## Active Radar Calibrator (ARC)



## The situation of external calibration

At JAXA tsukuba space center

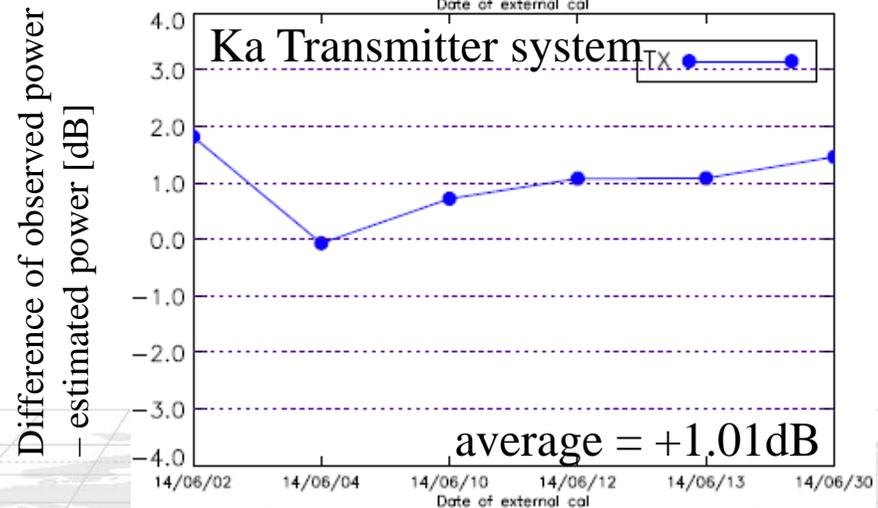
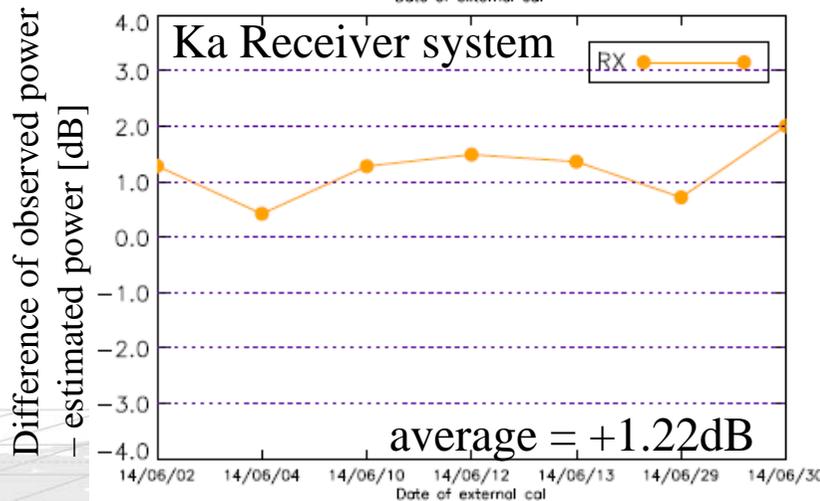
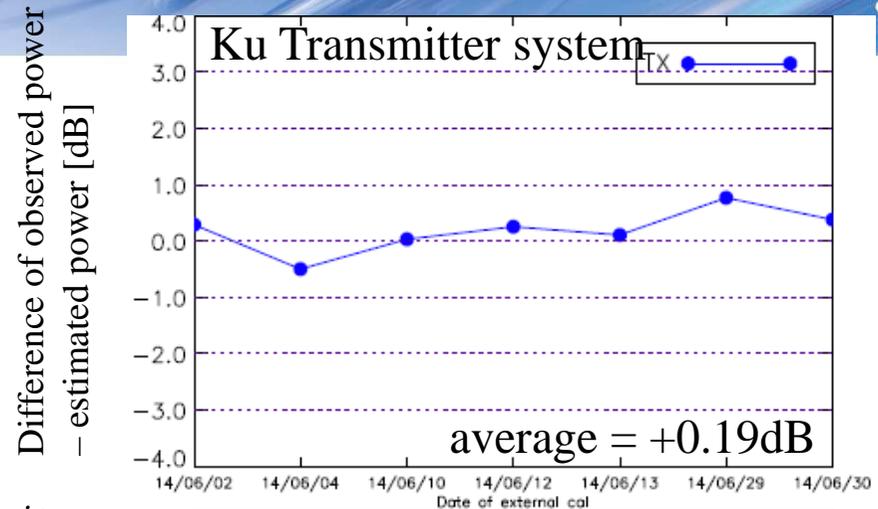
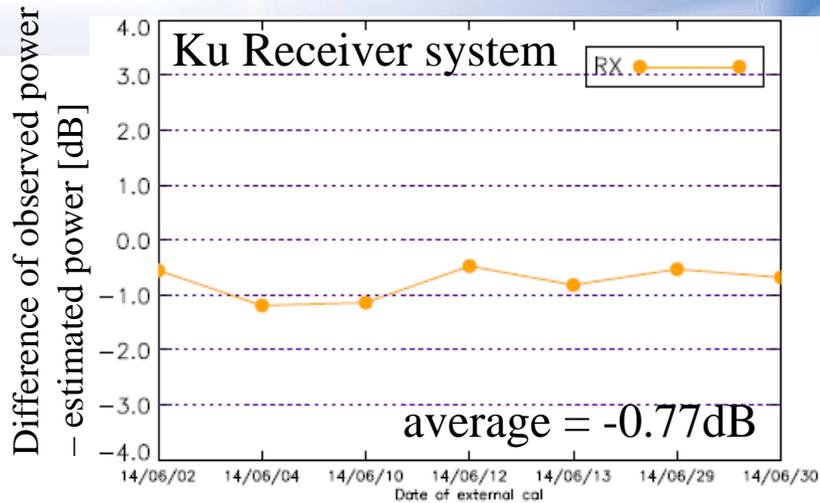


KaARC

Calibrator for  
the ARC

KuARC

# Results of External Calibration

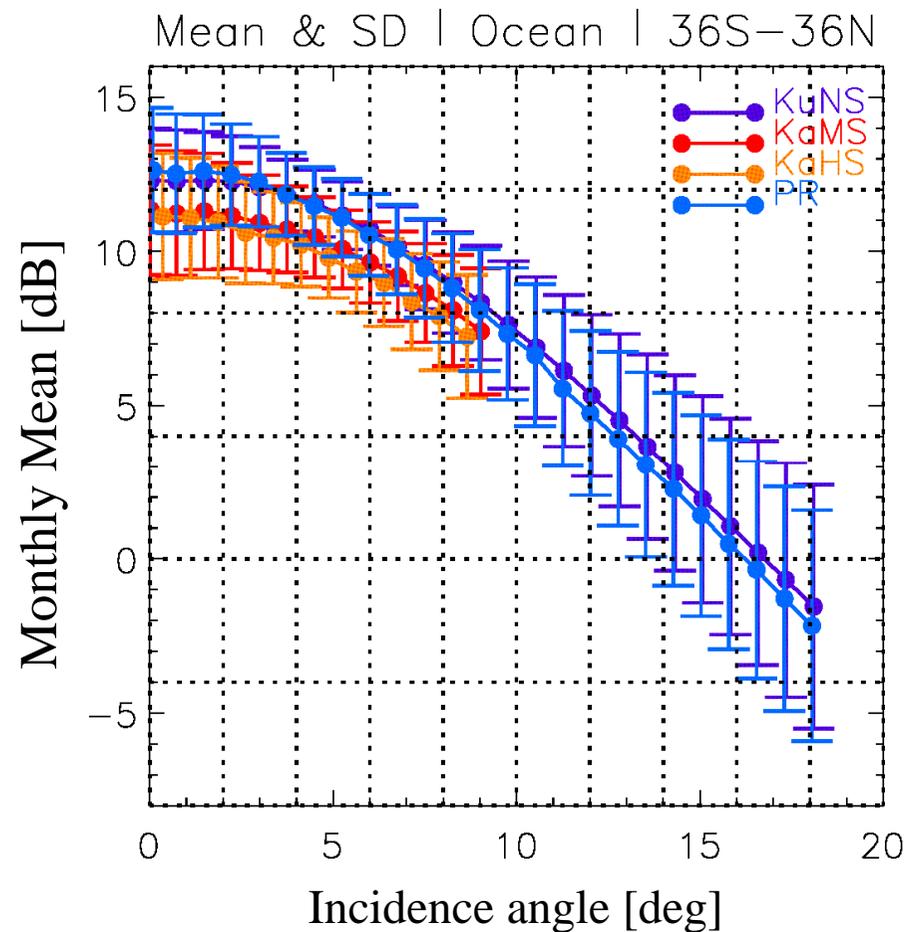


- Although the trends have some offsets, these are stable. JAXA will research for these offsets.
- Since the current calibration coefficient satisfies continuity of  $\sigma^0$  in DPR L2 from TRMM/PR (not shown), gain offsets found in external calibration has not been adapted now.

# Intercomparison Oceanic $\sigma^0$

- \* Comparison with reference to the TRMM/PR

Monthly mean & SD

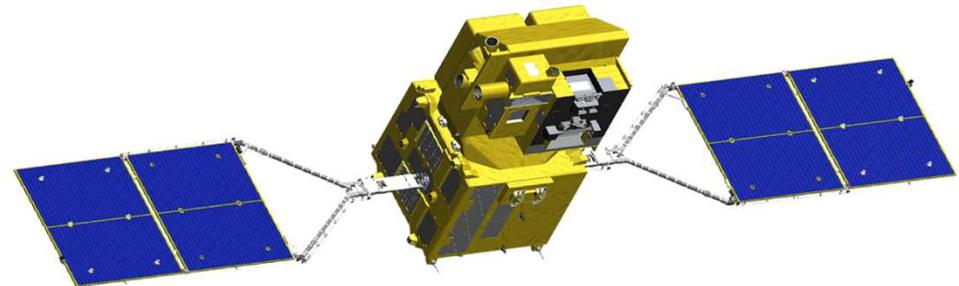
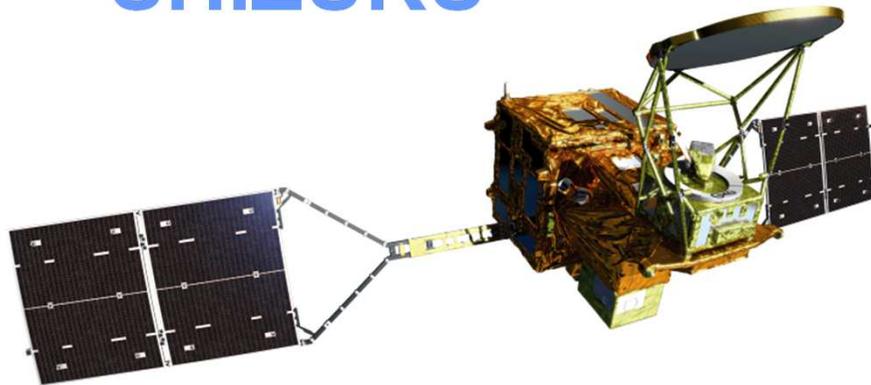


# GCOM 1<sup>st</sup> Generation Satellites



- 2 types of medium-sized satellites and 3 generations: 10-15 years observation

## “SHIZUKU”



### GCOM-W1 (Water)

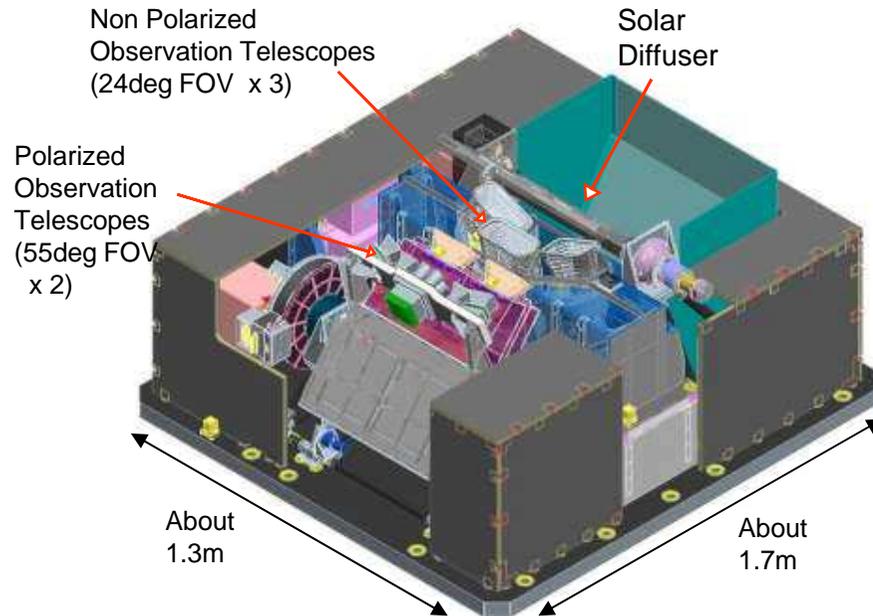
### GCOM-C1 (Climate)

Instrument	<b>Advanced Microwave Scanning Radiometer-2</b>
Orbit	Sun Synchronous orbit Altitude: 699.6km (on Equator) Inclination: 98.2 degrees Local sun time: 13:30+/-15 min
Size	5.1m (X) * 17.5m (Y) * 3.4m (Z) (on-orbit)
Mass	1991kg
Power gen.	More than 3880W (EOL)
Launch	May 18, 2012
Design Life	5-years

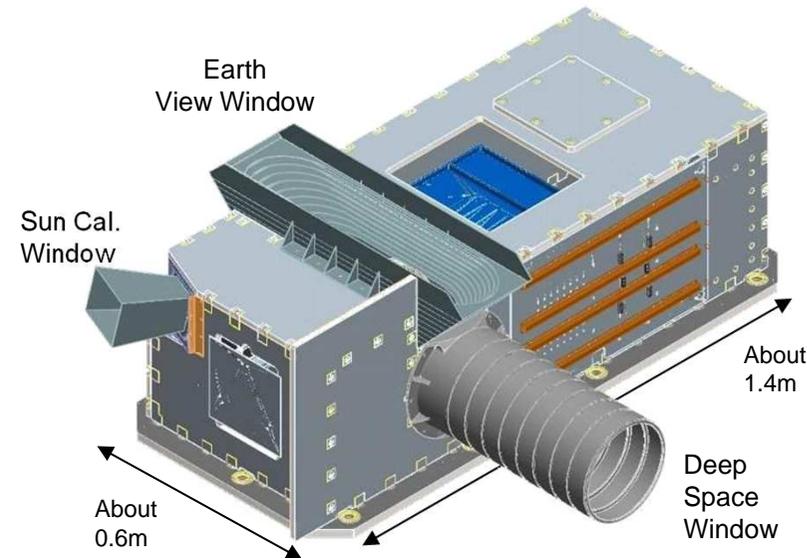
Instrument	<b>Second-generation Global Imager</b>
Orbit	Sun Synchronous orbit Altitude: 798km (on Equator) Inclination: 98.6 deg. Local sun time: 10:30+/- 15min
Size	4.6m (X) * 16.3m (Y) * 2.8m (Z) (on orbit)
Mass	2093kg
Power gen.	More than 4000W (EOL)
Launch	JFY 2016
Design Life	5-years



# SGLI on GC0M-C



Visible and Near Infrared Radiometer (SGLI-VNR)

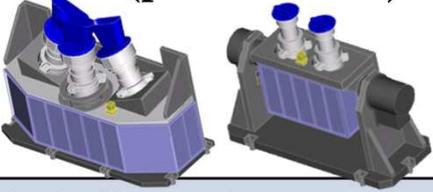
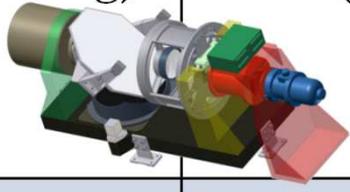


Infrared Scanning Radiometer (SGLI-IRS)

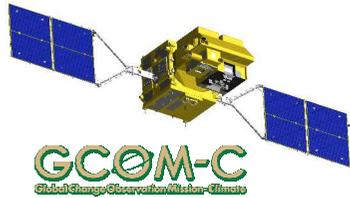
Sensor Unit	features
<b>SGLI VNR</b>	Non Polarized Observation (11ch), IFOV 250m, Swath 1150km Polarized Observation (2ch), IFOV 1km, Swath 1150km
<b>SGLI IRS</b>	Shortwave Infrared (SWI 4ch), IFOV 250m/1km, Swath 1400km Thermal Infrared (TIR:2ch), IFOV 500m, Swath 1400km



# SGLI calibration (radiometric)

Sensor component	VNR (push-bloom)	SWIR (scanning)	TIR (scanning)
Calibration target			
Pre-launch characterization	Gain (radiometric sensor model), diffuser, RSR, linearity, polarization, MTF (PSF), ..		Gain, blackbody, RSR, linearity..
Functional check	Electric calibration, sensor telemetry monitoring..		
Offset	<b>Optical black</b> (every line), and nighttime observations	<b>Deep-space</b> or nighttime observations	<b>Deep-space</b> and pitch maneuver observations
Launch shift	<b>LED</b> to check change of <b>diffuser</b>	halogen lamp (+LED) to check diffuser change	
Short term gain change	solar light → <b>diffuser</b> (~once/week)	solar light → light guide → <b>diffuser</b> (every path)	<b>Black body</b> calibration
Long term change	Monthly <b>Moon</b> (7°) observations by pitch maneuver for evaluation of the diffuser degradation ( <i>with GSICS</i> )		<i>Primary source</i>
Vicarious adjustment	Vicarious calibration over the CEOS instrumented sites and ocean cruises	Vicarious calibration at the CEOS instrumented sites (land)	
Cross check and image quality	Vicarious & cross calibration over the CEOS invariant sites (Libya, Dome-C, TuzGolu..), stray light by moon, pol sensitivity by simultaneous VNR-PL..		

*These tasks will be led by the joint team of the JAXA GCOM-C hardware development and data-analysis & application groups*

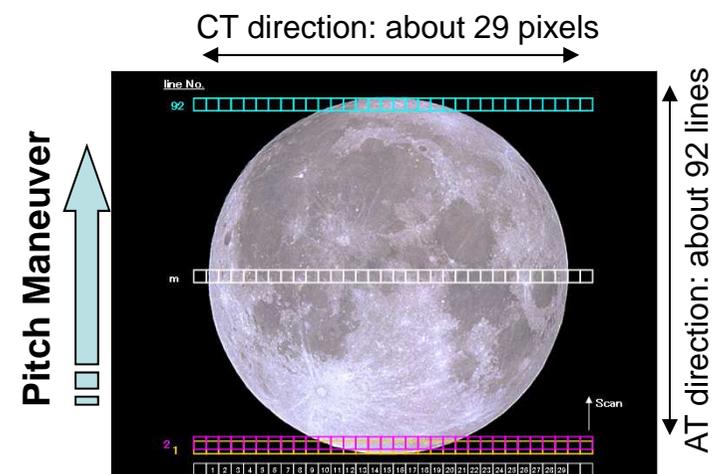
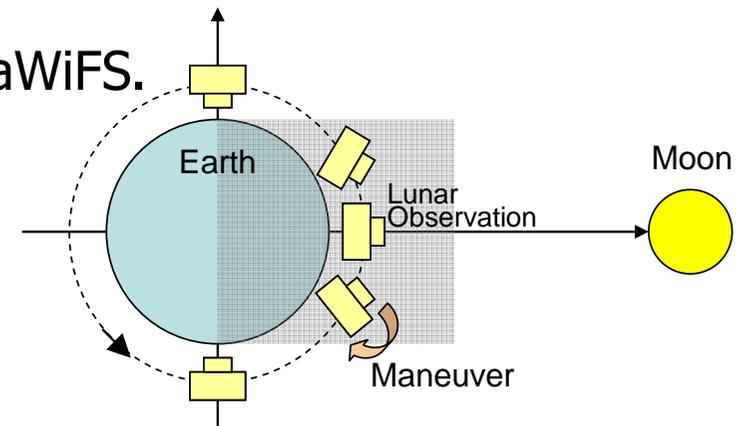


# SGLI Lunar Calibration



- Moon reflecting solar light is a stable light source as a long term calibration reference of the optical sensors.
- GC0M-C lunar calibration maneuvers are planned to be conducted every 29.5 days during 5 years mission.
- Lunar calibration concept is similar to SeaWiFS.

Calibration interval	Every 29.5 days (= synodic period of the moon and the sun)
Lunar phase angle	7deg +/-3deg
SGLI lunar observation	All bands (VNR & IRS) 250m resolution
Satellite Maneuver Requirement	- Pitch rate of 0.15 deg/s with high stability - Selectable roll angle (lunar image in SGLI swath)





# TANSO-FTS and CAI specifications

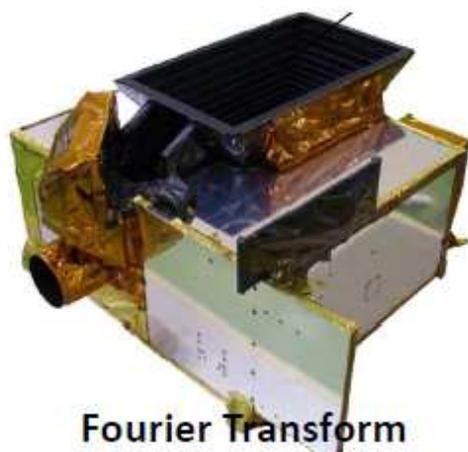


2009 ~ present in operation



GOSAT

Thermal And Near infrared Sensor for carbon Observation (TANSO)



Fourier Transform Spectrometer (FTS)



Cloud and Aerosol Imager (CAI)

## Fourier Transform Spectrometer (FTS)

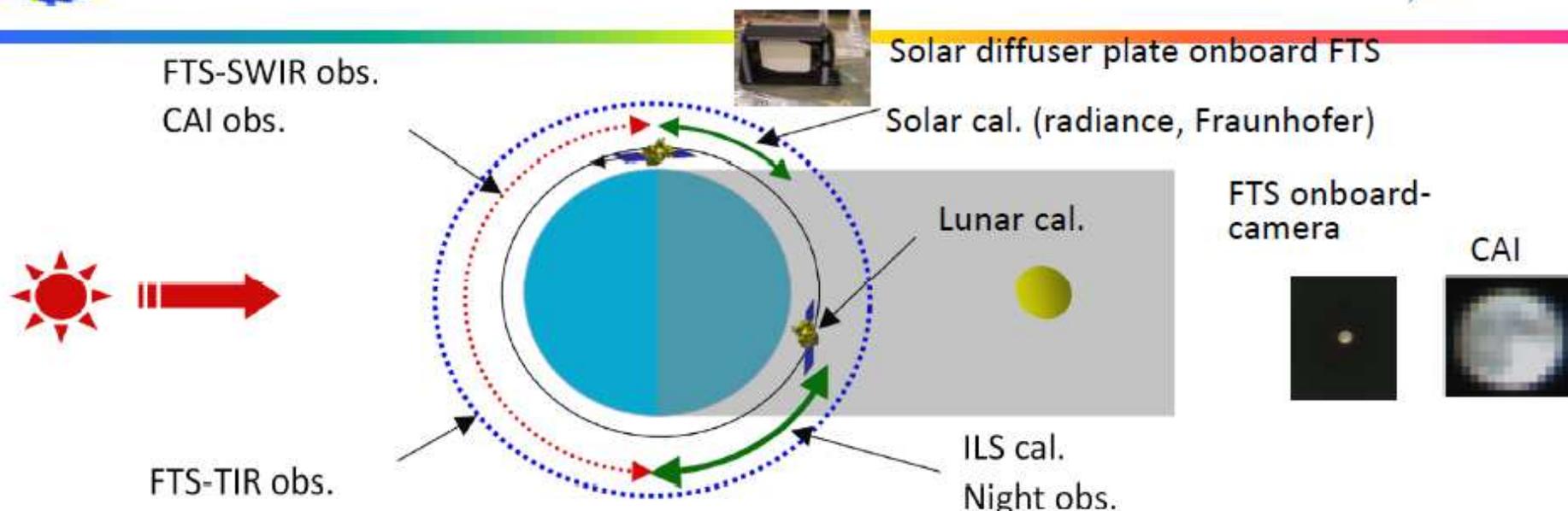
Mission	GHGs measurements
Band	SWIR-0.76 $\mu$ m, 1.6 $\mu$ m, 2.0 $\mu$ m bands with P/S polarization (O <sub>2</sub> -A, CO <sub>2</sub> , CH <sub>4</sub> , H <sub>2</sub> O band)
	TIR-5.5~14.3 $\mu$ m (CO <sub>2</sub> , CH <sub>4</sub> , O <sub>3</sub> band)
Spec. Res.	0.2cm <sup>-1</sup>
Swath	750km ex: 5 points / every 180km
Footprint	10.5km

## Cloud and Aerosol Imager (CAI)

Mission	Cloud detection and aerosol correction within FTS IFOV
Band	0.38, 0.67, 0.87, 1.60 $\mu$ m band
Swath	750-1000km
Footprint	0.5 and 1.5km



# GOSAT calibrations



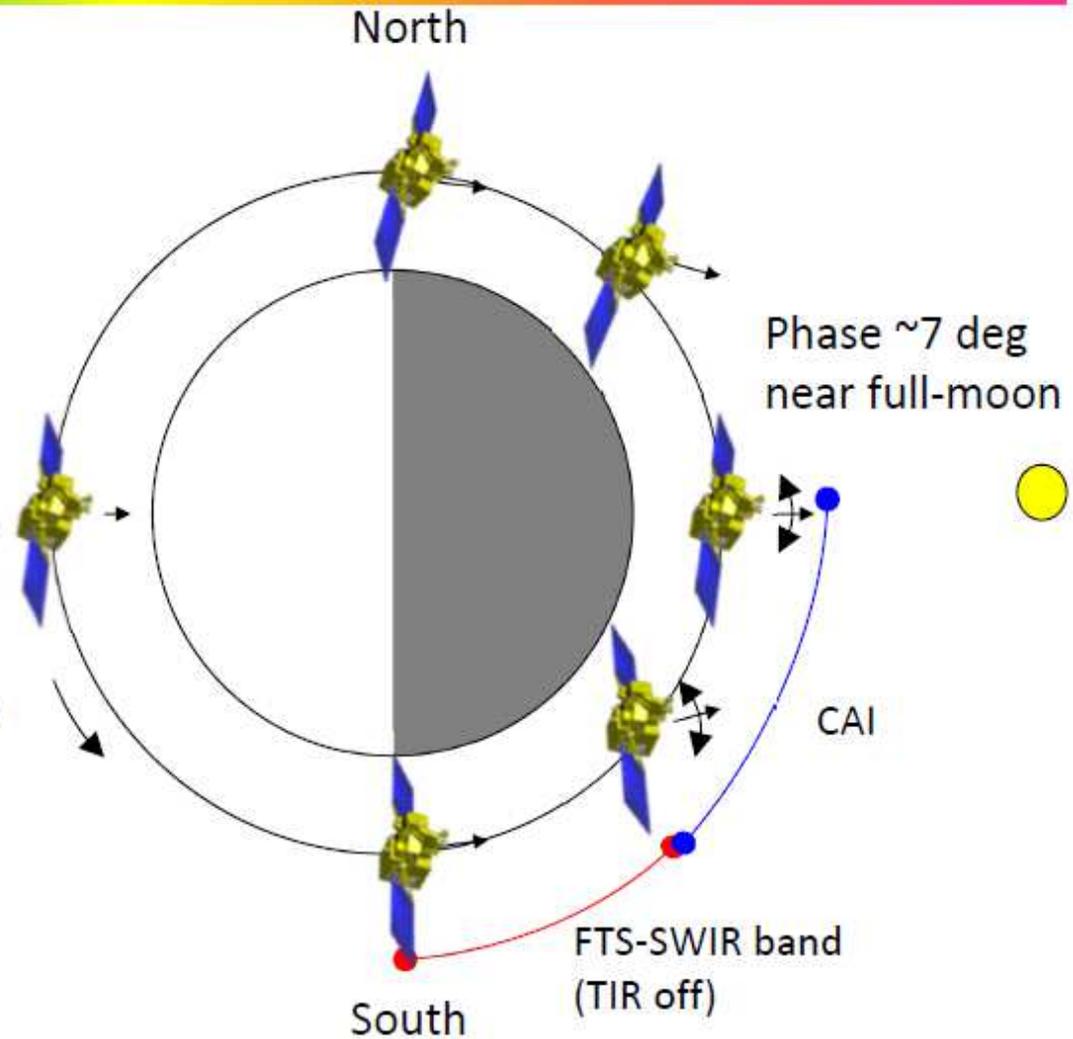
	FTS				CAI			
	B1P/S 0.76um	B2P/S 1.6um	B3P/S 2um	B4 5.5-14.3um	B1 0.38um	B2 0.67um	B3 0.87um	B4 1.6um
Radiance	Vicarious calibration (1/year) Solar calibration (backside: 1/month) <b>Lunar calibration (2/year)</b>			Vicarious calibration (1/year) Cross comparison (AIRS)	Vicarious calibration (1/year) <b>Lunar calibration (2/year)</b> Dark (Night observation: 1/month)			
Spectral features	Fraunhofer line (spectral shift) ILS calibration (B2P/S: 1/month)			-	-			
Geometry	FTS onboard-camera image is validated by using AVNIR-2.				Validated by using GSHHS			



# Lunar calibration for GOSAT

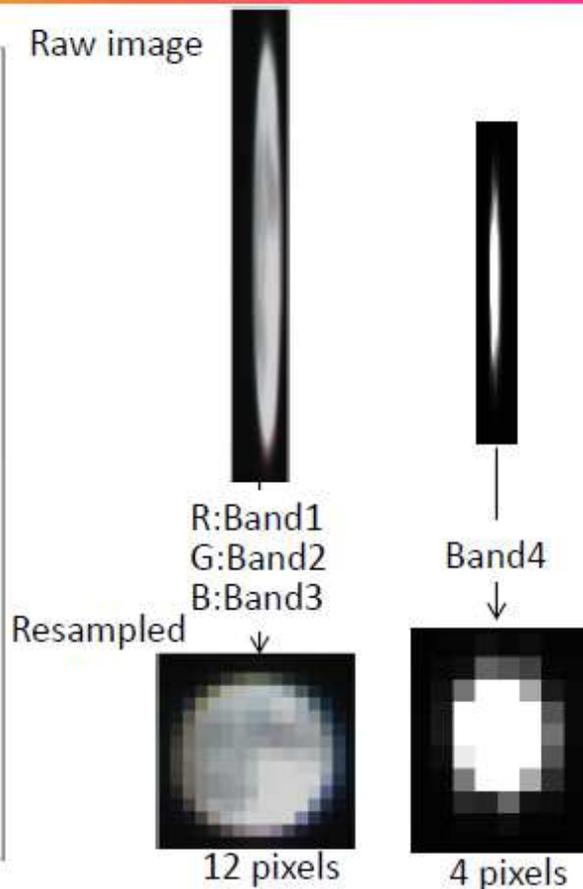
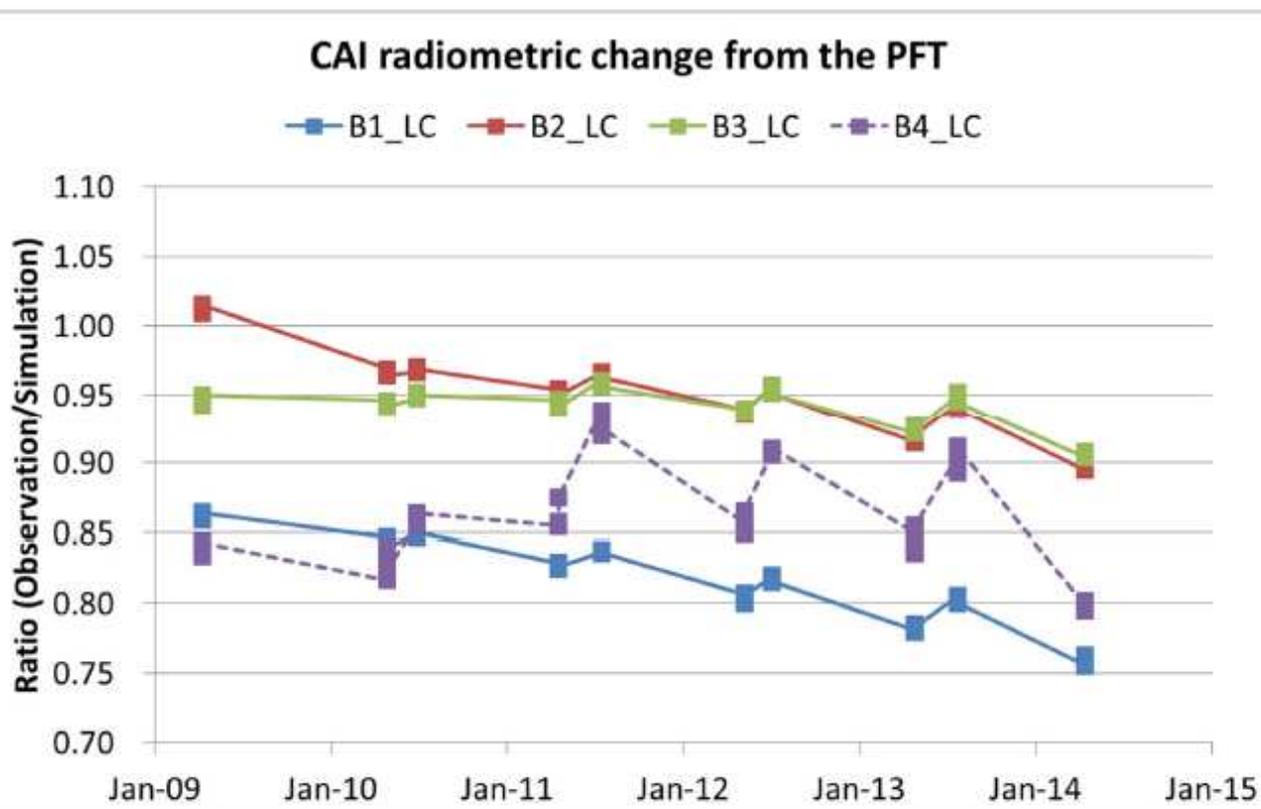


- Radiometric onboard calibration for FTS SWIR and CAI
- For FTS, gazing the moon by the satellite pointing with half IFOV
- For CAI, scanning the moon by the satellite pitch motion
- Once a year (also with backup, i.e. total twice)
- Bright and stable target with observation phase angle of 7 degrees near full-moon





# CAI lunar calibration result



- Band4 IFOV is broader than the identical. It might not be well-evaluated.
- Band1-3 radiometric trends are evaluated well.

CAI lunar observation on 28 April 2010. Images are oversampled in along-track direction.

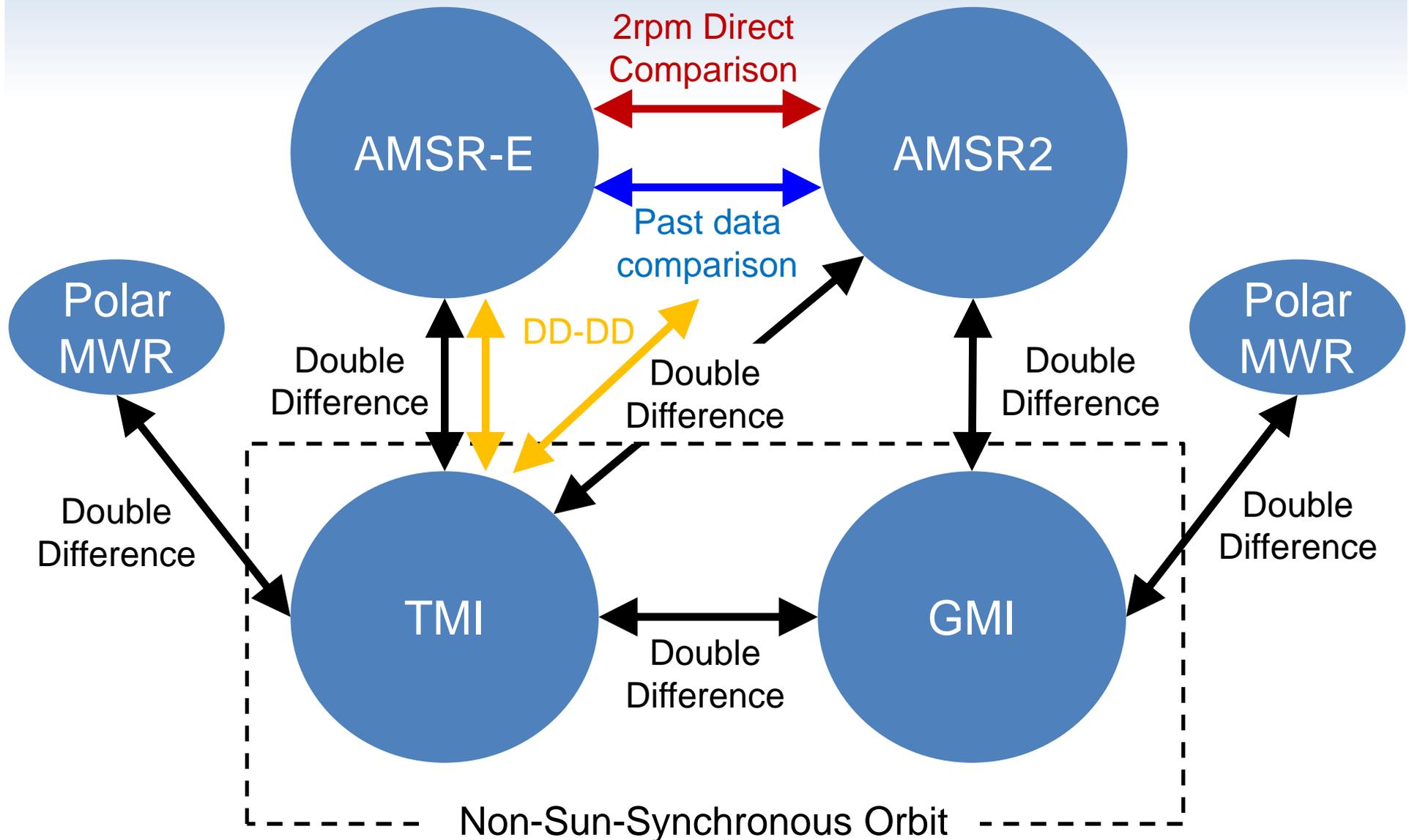
# MWR for Intercomparison

	<b>TMI (TRMM)</b>	<b>AMSR-E (Aqua)</b>	<b>AMSR2 (GCOM-W)</b>	<b>GMI (GPM)</b>
<b>Launch Year</b>	<b>1997</b>	<b>2002</b>	<b>2012</b>	<b>2014</b>
<b>Sensor appearance</b>				
<b>Channels for Intercalibration</b>	10.65 V/H 19.35 V/H 21.3 V 36.5 V/H 85.5 V/H	10.65 V/H 18.7 V/H 23.8 V 36.5 V/H 89.0 V/H	10.65 V/H 18.7 V/H 23.8 V 36.5 V/H 89.0 V/H	10.65 V/H 18.7 V/H 23.8 V 36.64 V/H 89.0 V/H
<b>Approximate incidence angle [degree]</b>	53.4 * (after boost)	55.0 54.5 for 89B	55.0 54.5 for 89B	52.8
<b>IFOV at 36GHz [km]</b>	16*9	14*8	12*7	16*9

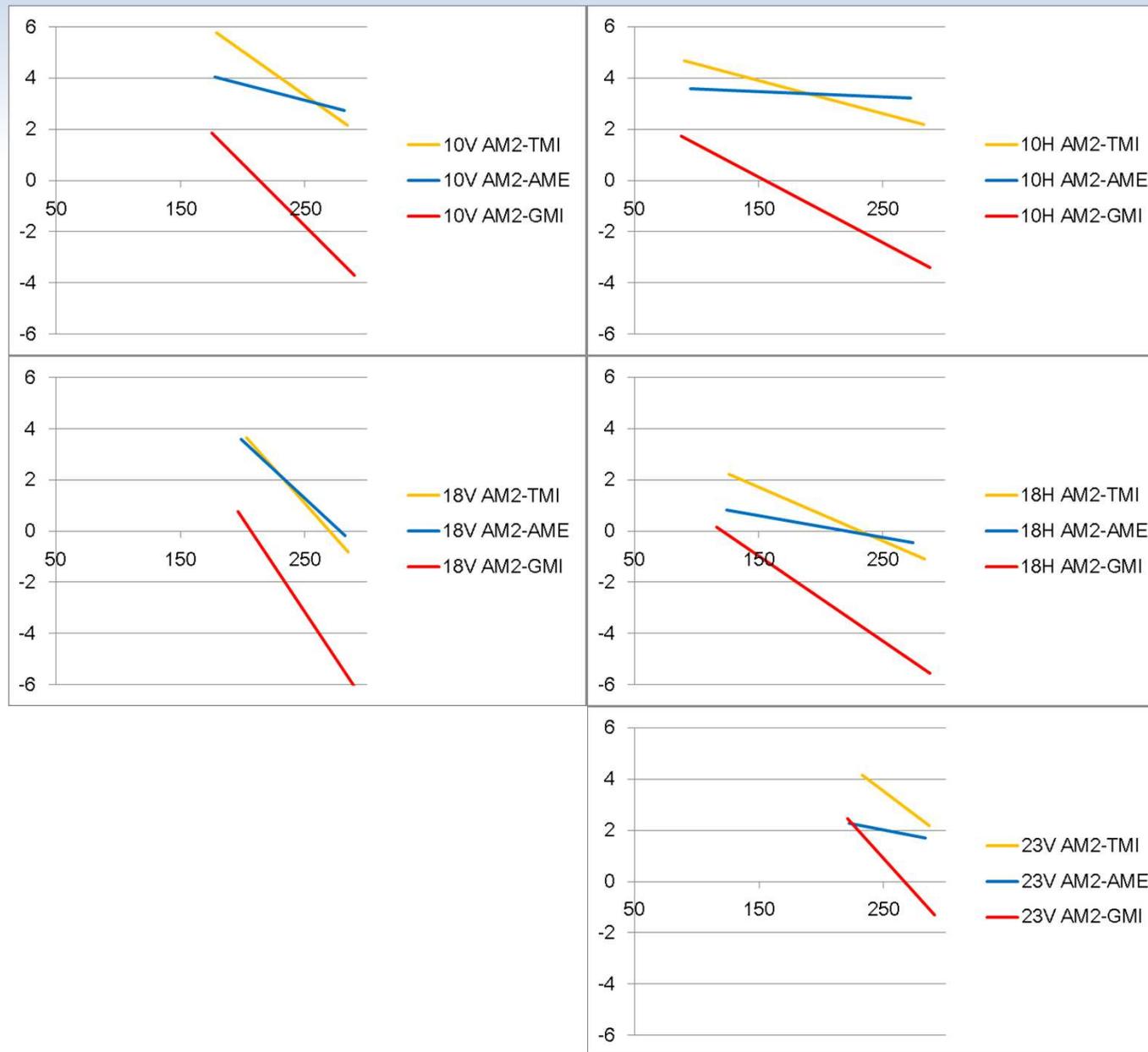
\* Recent information on TMI incidence angle through GPM X-CAL team: Incidence angles at 10V, 10H, and 37V are differ from the base value about 0.65, 0.22, and 0.12 degrees, respectively.

*Photo of TMI and GMI by NASA*

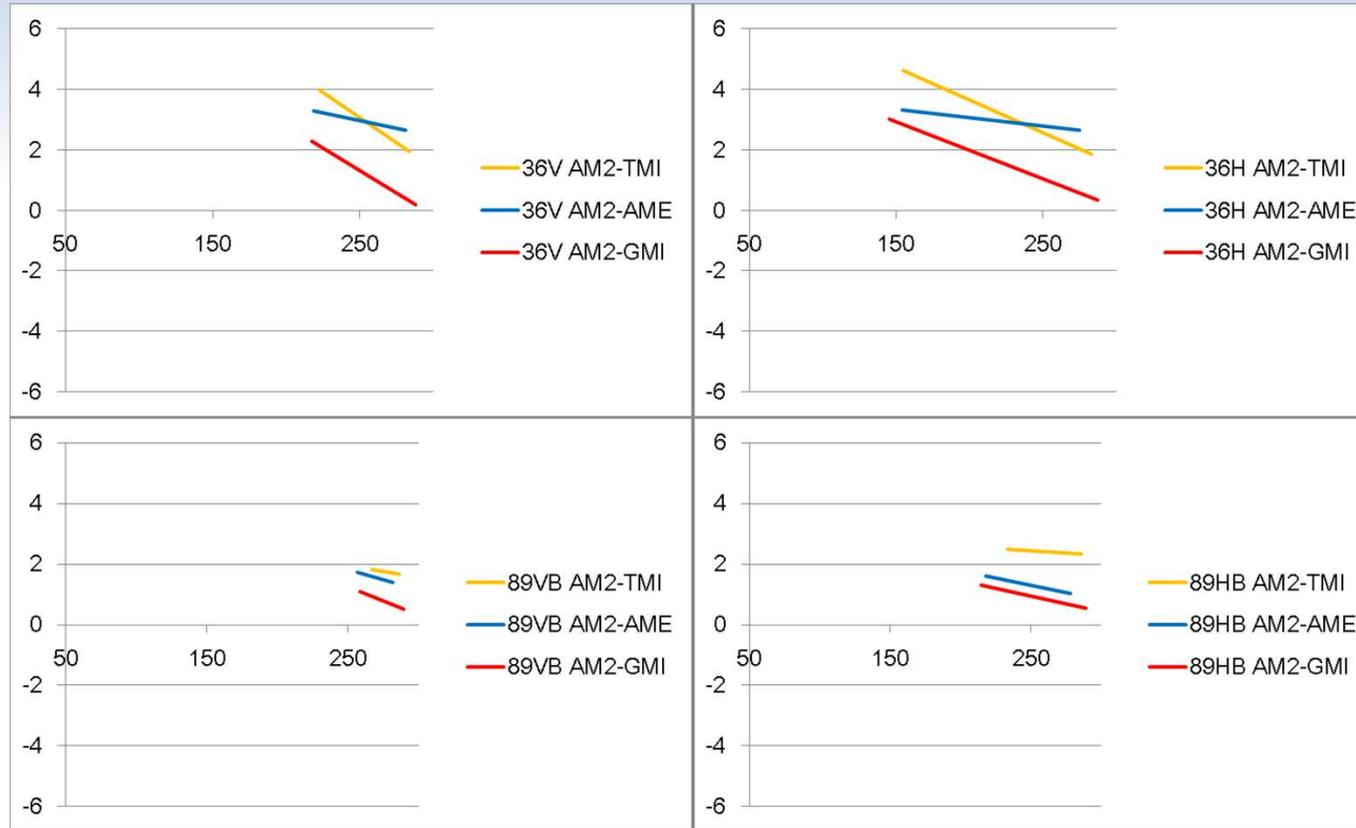
# Overview of Intercomparison



# Intercomparison Summary



# Intercomparison Summary



Thank you