

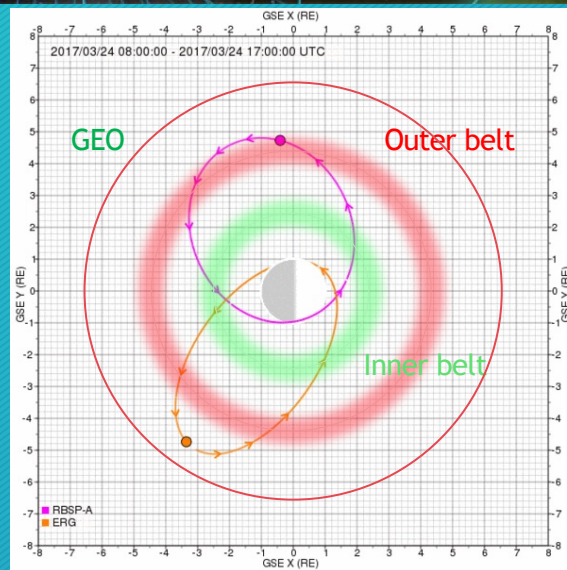
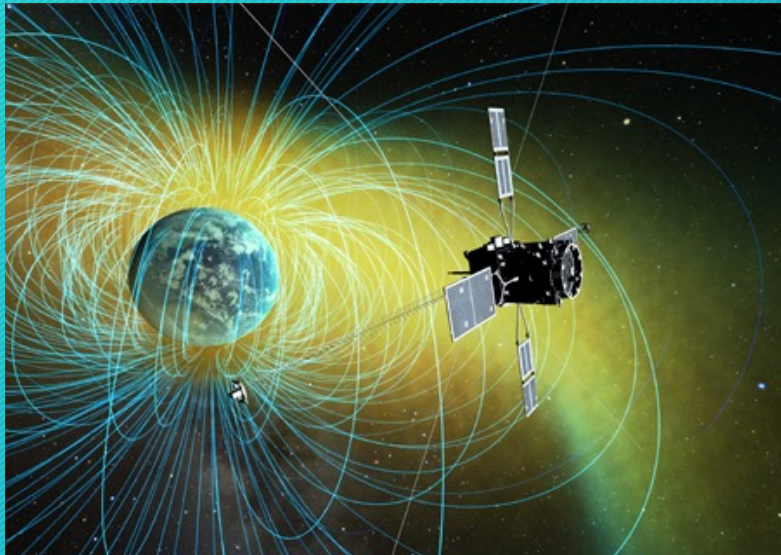
# Calibration of radiation belt electron data observed by Arase satellite using Geant4 simulation

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Space weather laboratory / Researcher - Park Inchun

2024. 3. 12. GSICS meeting

# ARASE satellite (ERG mission)

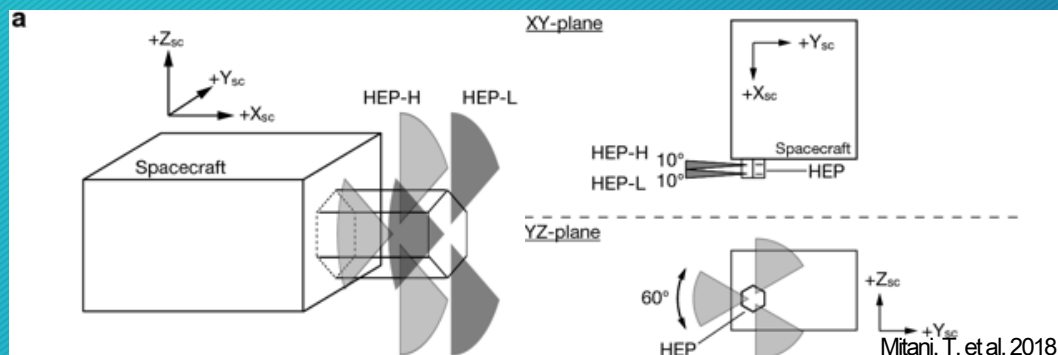
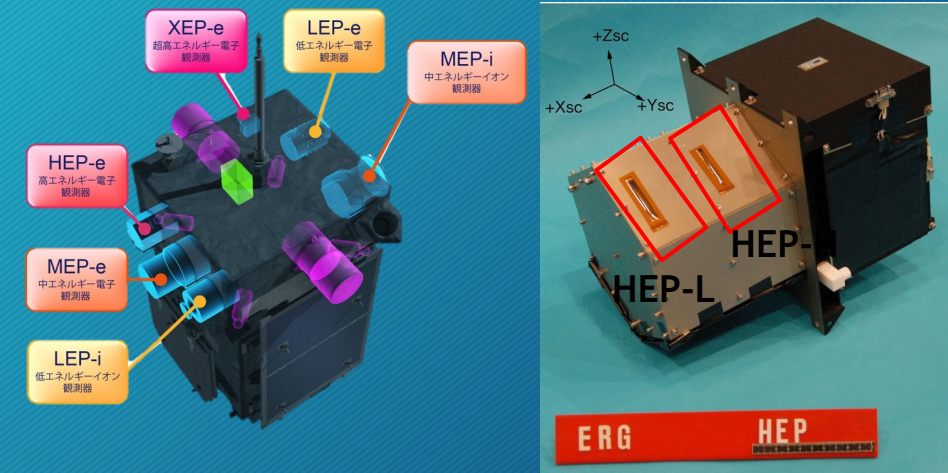


Launch	Date	20 December 2016 (Observation start: March 2017)
	Location	Uchinoura Space Center (USC)
	Launch Vehicle	Epsilon
Nominal Mission Life		>1 yr
Orbit	Altitude	Perigee: about 460 km, Apogee: about 32,110 km
	Inclination	31°
	Type of Orbit	Elliptical orbit
	Period	about 9 hours (565 min)
Attitude	Stabilization	Spin-stabilized
	Spin Direction	Sun-oriented
	Spin period	8 s (7.5 rpm)
Satellite Bus		SPRINT bus
Configuration	Weight	350 kg

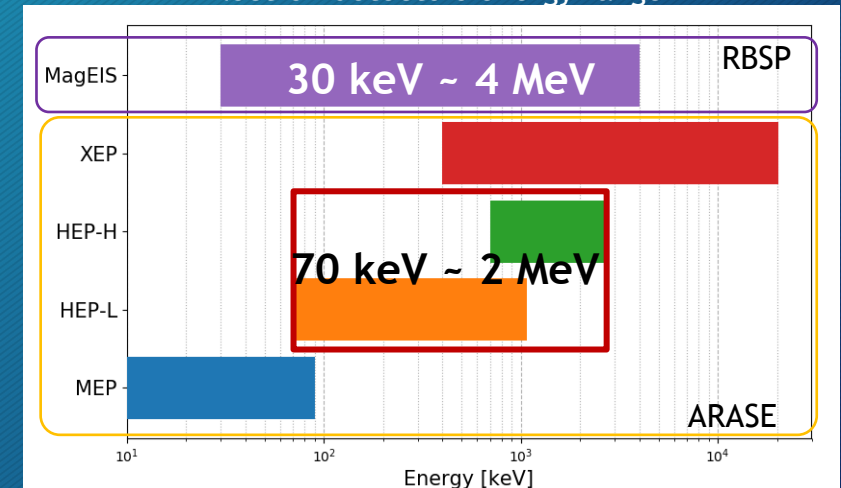
# ARASE instruments



- Low-Energy Particle Experiments - Electron Analyzer (LEP-e)
- Low-Energy Particle Experiments - Ion Mass Analyzer (LEP-i)
- Medium-Energy Particle Experiments - Electron Analyzer (MEP-e)
- Medium-Energy Particle Experiments - Ion Mass Analyzer (MEP-i)
- **High-Energy Electron Experiments (HEP-L, HEP-H)**
- Extremely High-Energy Electron Experiments (XEP)
- Plasma Wave Experiment (PWE)
- Magnetic Field Experiment (MGF)
- Software-Type Wave Particle Interaction Analyzer (S-WPIA)



Electron detectors energy range



# Calibration of radiation belt electron data observed by Arase satellite using Geant4 simulation

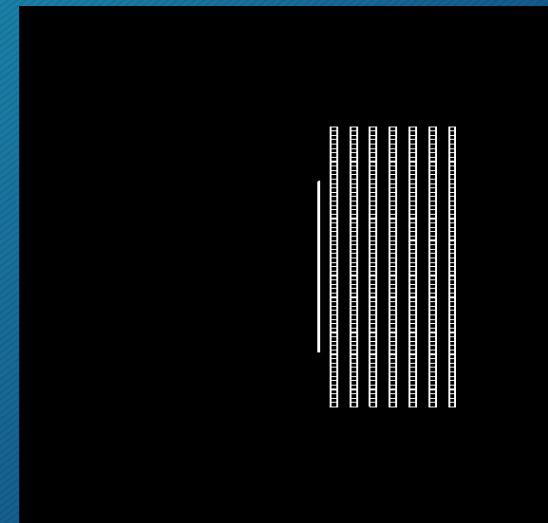
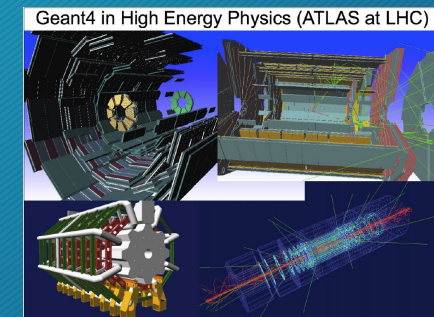


## ➤ Purpose of the Study

- In-flight ARASE HEP instrument data calibration and performance verification

## ➤ What is Geant4?

- Geant4 (for GEometry ANd Tracking )
- Developed by CERN, INFN, KEK, SLAC (ver1 released 1994, now 11.1)
- Monte Carlo simulation tool for particle physics
- Simulate interactions when elementary particles passing through and interacting with matter



# Data Calibration Process



✓ Raw data: HEP electron count rate [Counts/s]



- ✓ Calibrate the count rate depending on the Energy
- Restore electron energy changed by AI window, structure factors
    - Using simulated response function



- ✓ Convert to Differential flux [Counts/mm<sup>2</sup> · str · keV · s]
- Depends on detector's characteristics
    - G-factor [mm · str] : change count rate to differential flux

Geant4 simulation

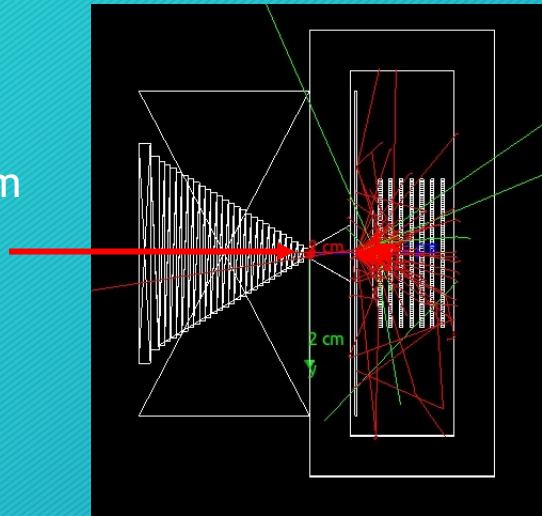
Lv2 Data Produce



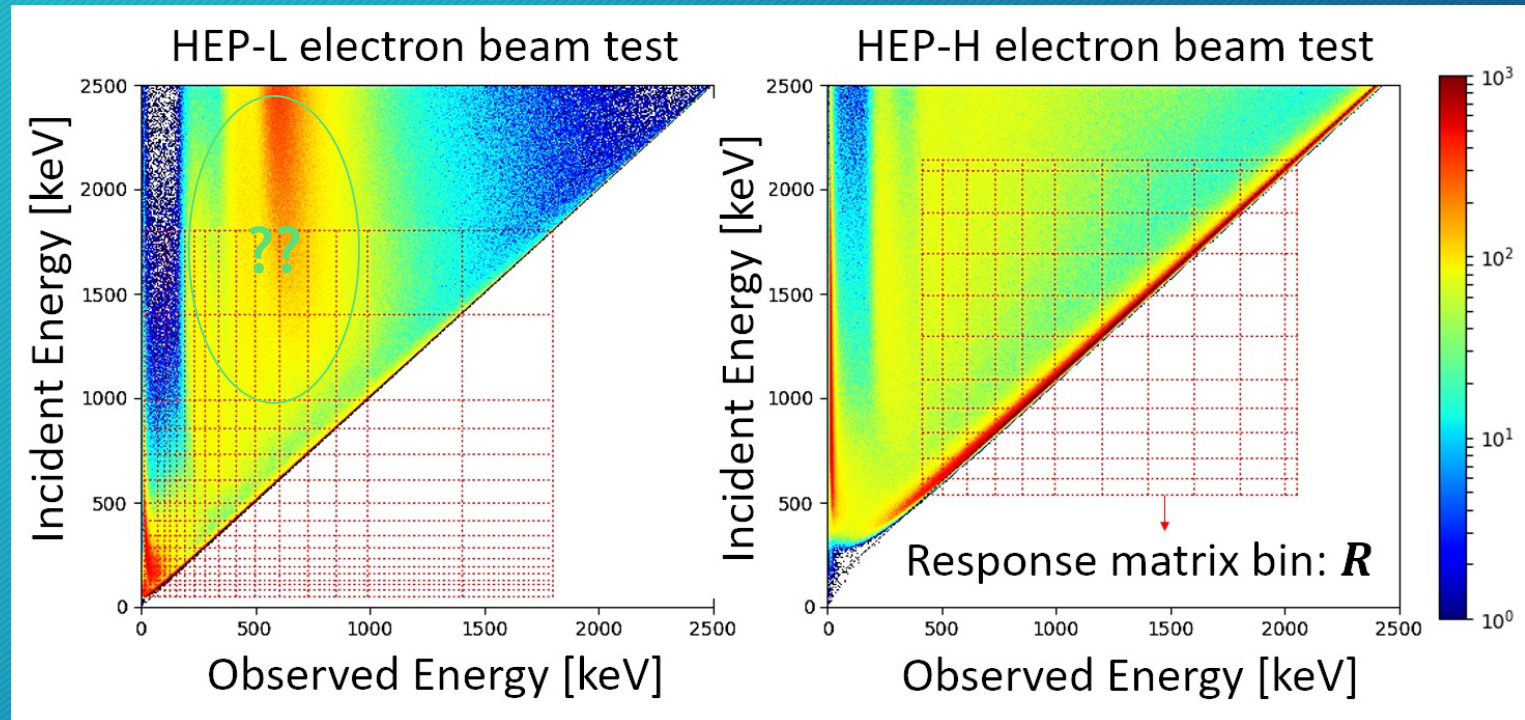
# Calibrate the count rate depending on the Energy

## <Simulation status>

$e^-$  Pencil beam  
1~3000 keV  
 $10^5$  events



## <Simulation results>

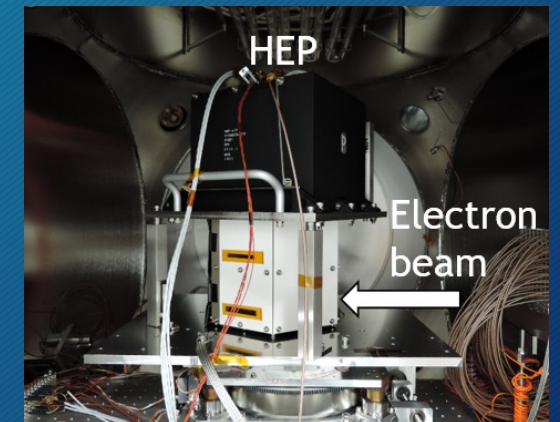
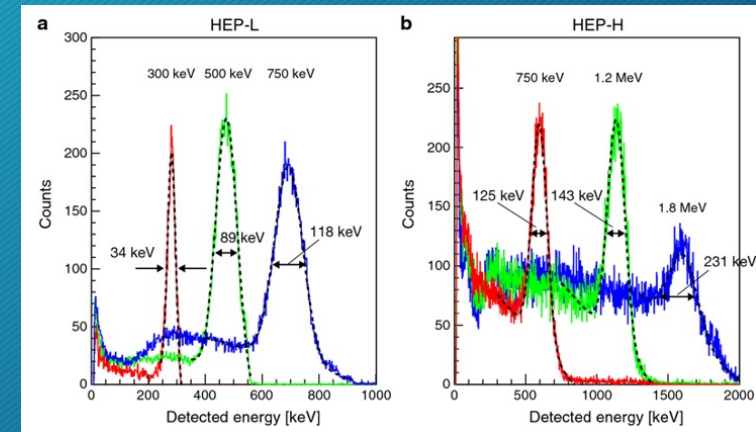


# Calibrate the count rate depending on the Energy



- Spectrum structure analysis
  - Beam test and simulation have dent structure
  - Thin and long SSD structure makes loss detection area
  - Energy deposit depends on thickness of loss detection area (scatter angle  $> 60^\circ \rightarrow$  Energy deposit  $> 450\text{keV}$ )
- Identify the causes of spectral changes
  - ➔ Instrument structure affects the Spectroscopic performance

## <Electron Beam test>



# Calibrate the count rate depending on the Energy

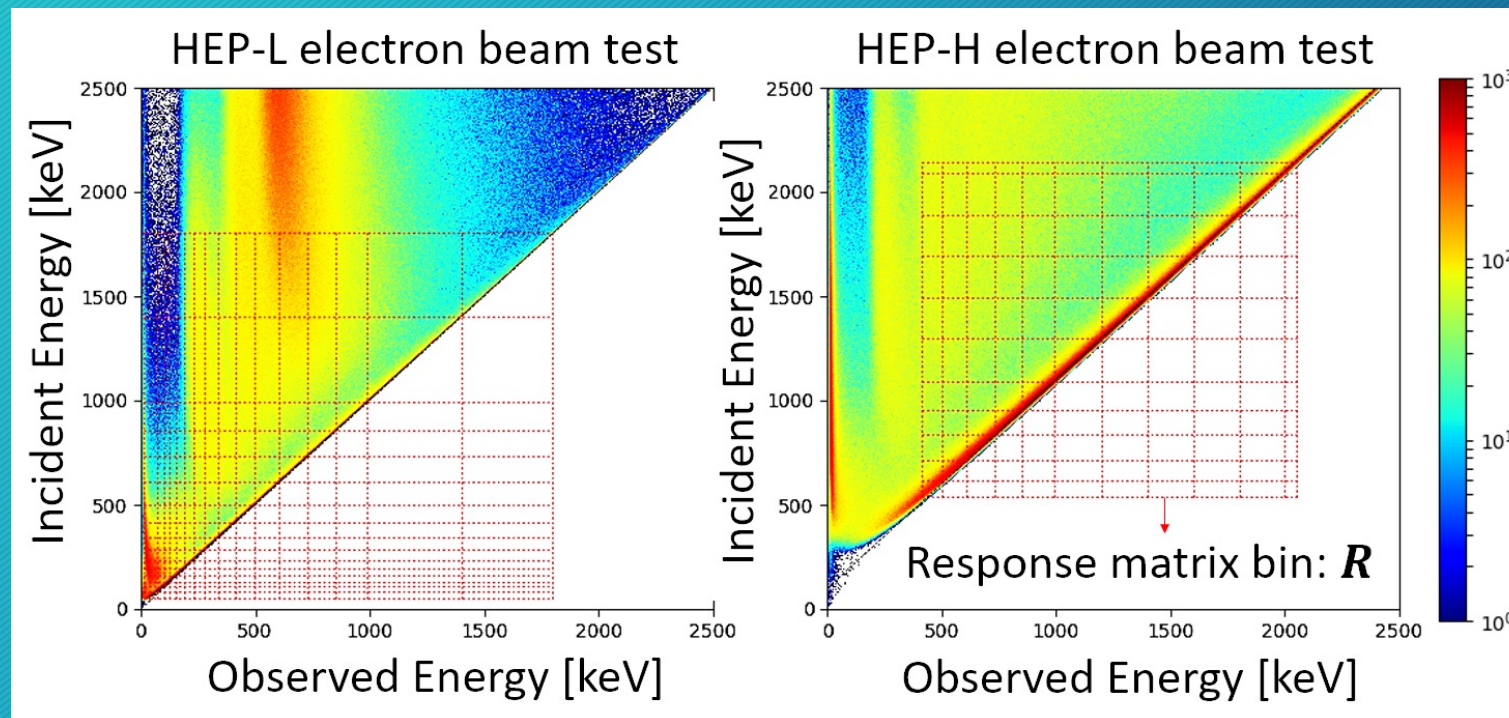
- Using the inversion matrix method, incident energy spectra restored from observed particle counts.

$$C_{true}(E_i) = R^{-1} \cdot C_{obs}(E_o)$$

※  $R$ : Response function

$C_{true}(E_i)$ : Energy spectra of incident electron counts

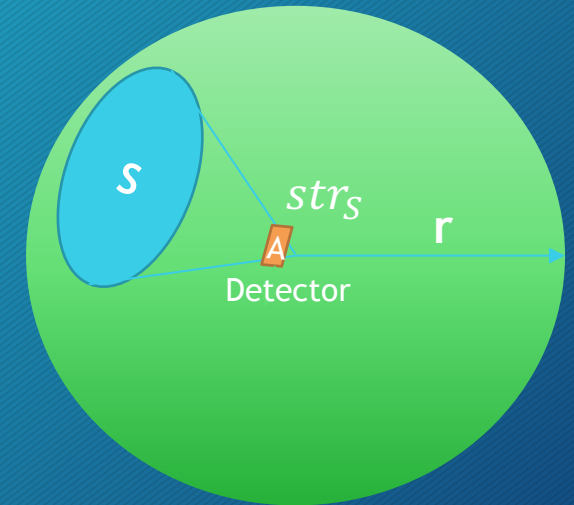
$C_{obs}(E_o)$ : Energy spectra of observed electron counts





# Convert to Differential flux (G-factor)

- G-factor?
  - The constant of detector to calculate differential flux.
  - FOV of detector( $str_S$ ) x Detector surface area(A)  
Pre cal HEP-H G-factor : 0.037 cm<sup>2</sup>str



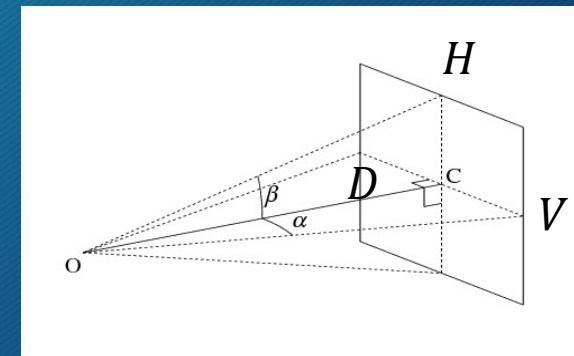
- How to calculate  $str_S$ ?

- 1. From the formula
  - Using FOV  $str_S = 4 \sin^{-1}(\sin \alpha \sin \beta)$
  - Using collimator geometry(Pre cal method)

$$str_S = 4 \cos^{-1} \sqrt{\frac{1 + \left(\frac{V}{2D}\right)^2 + \left(\frac{H}{2D}\right)^2}{\left(1 + \left(\frac{V}{2D}\right)^2\right) \left(1 + \left(\frac{H}{2D}\right)^2\right)}}$$

- 2. Using simulation

$$str_S = \frac{S}{r^2}$$

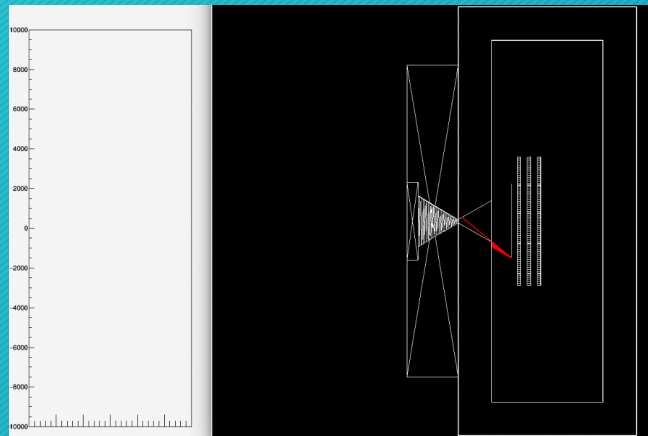


# Convert to Differential flux (G-factor)

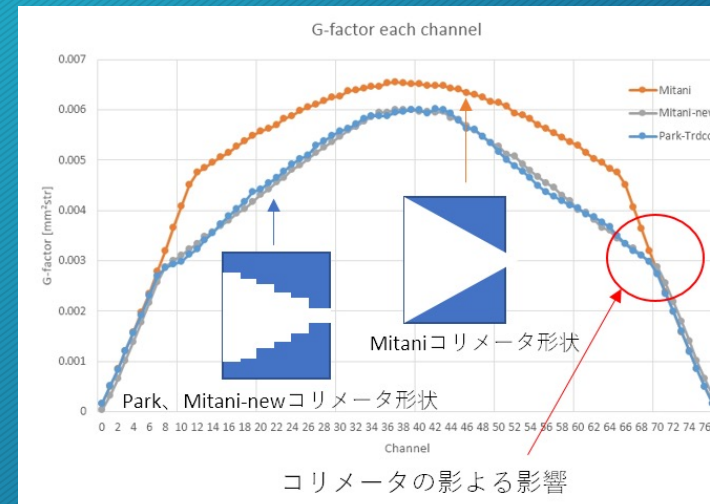
- Calculation of G-factor from the simulation

$$G\text{-factor} \simeq 2\pi \times \text{Sensor area} \times \frac{\text{Escaped particle}}{\text{Irradiated particle}} \text{ [mm}^2\text{sr]}$$

HEP-L: ~0.0031 mm<sup>2</sup>sr / HEP-H: ~0.031cm<sup>2</sup>sr



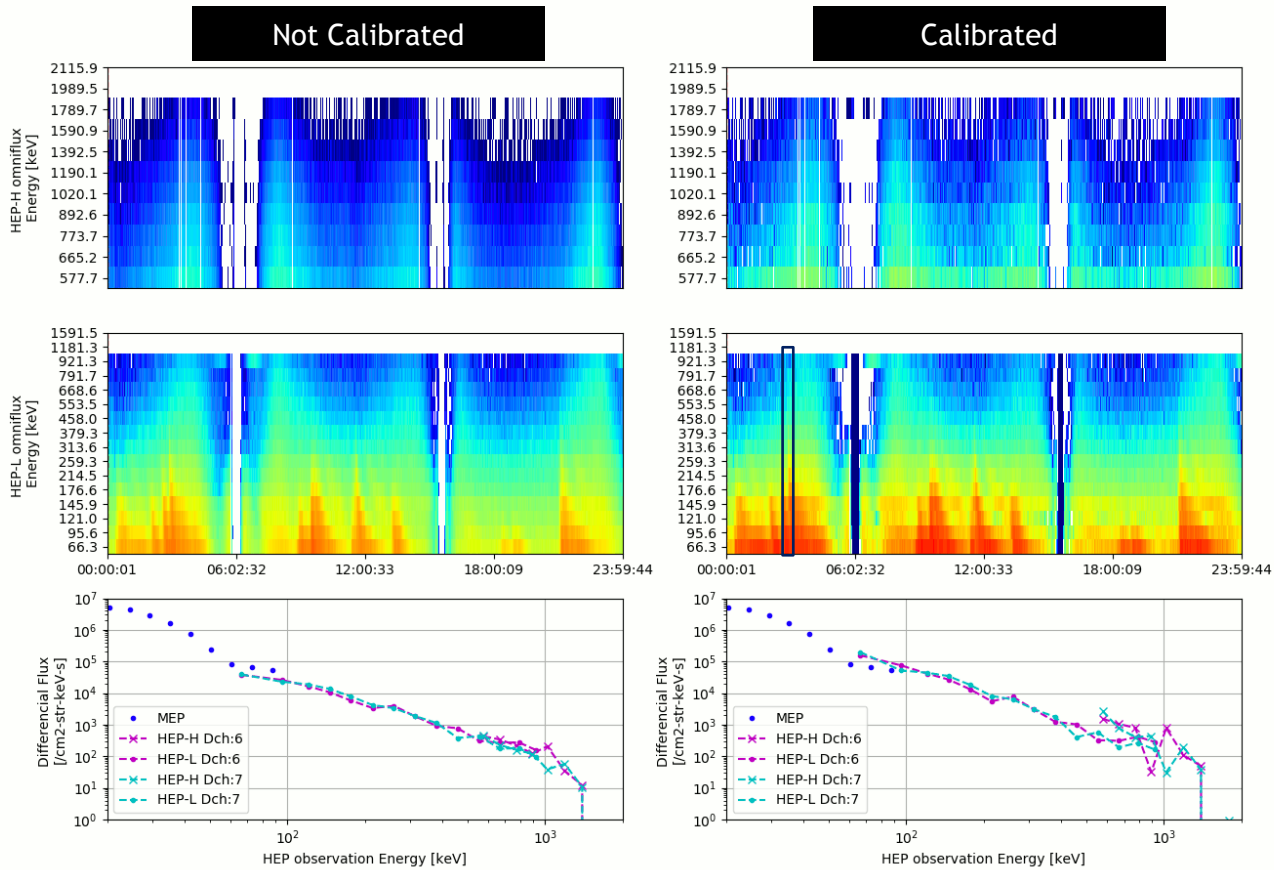
<15m 離れた壁に投影した影>



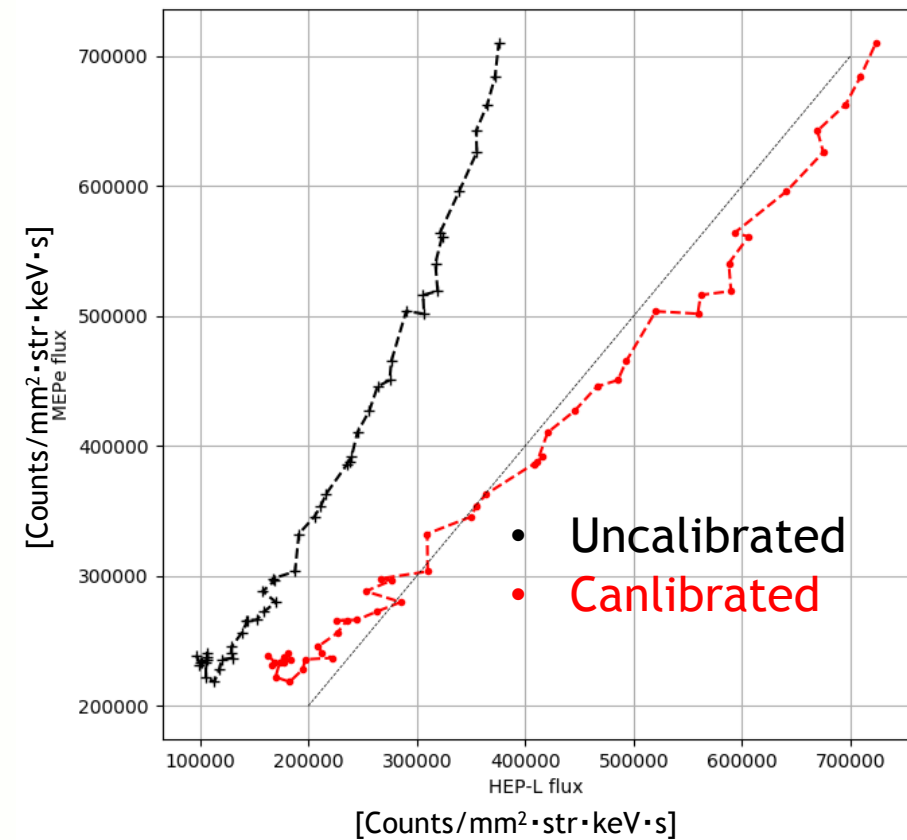
コリメータの影による影響

✓ Simulation founded that the detailed geometry affect the result (Collimator shape, slit thickness)

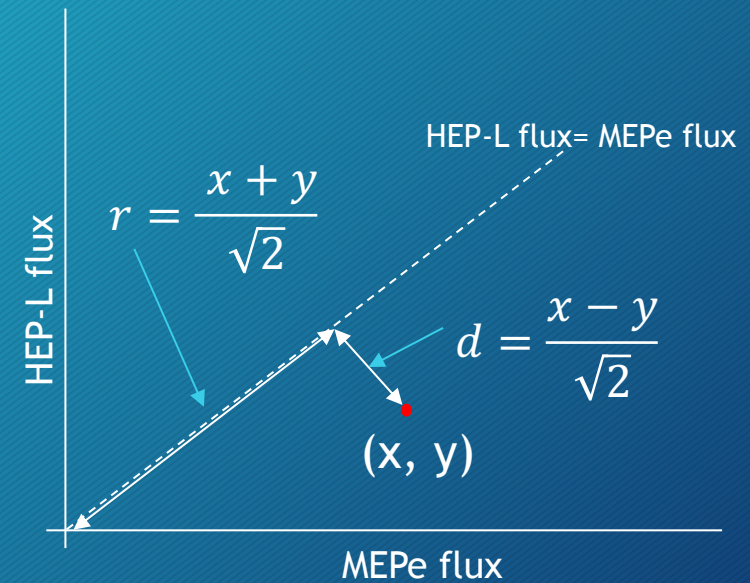
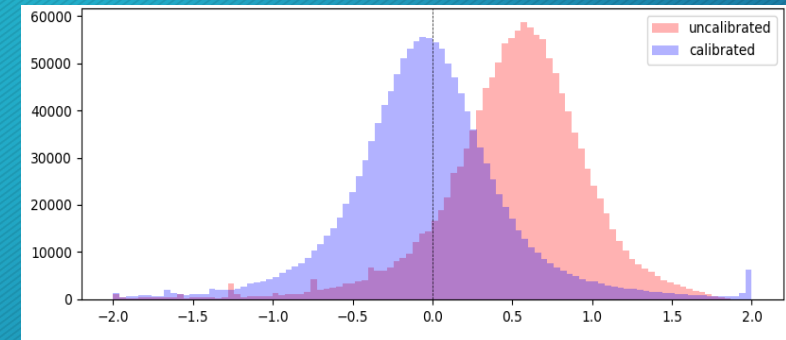
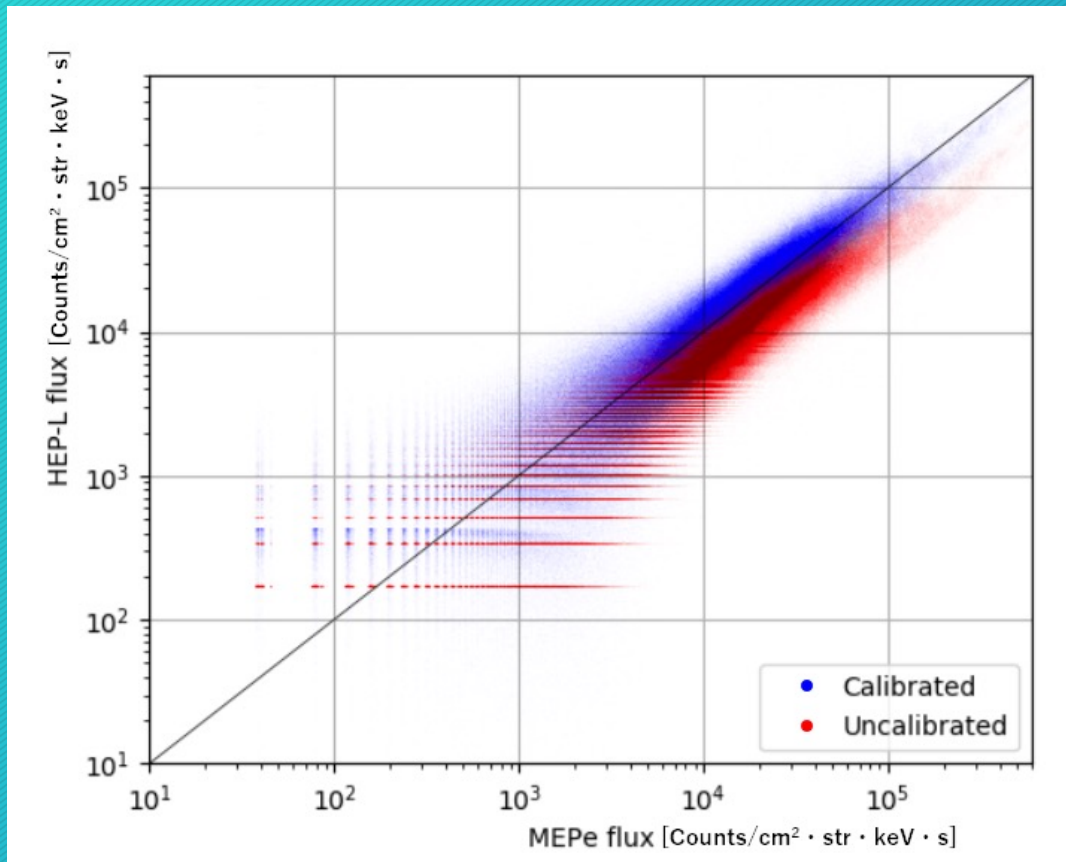
# Calibration results(1 day)



## 95 keV electron MEPe:HEPe

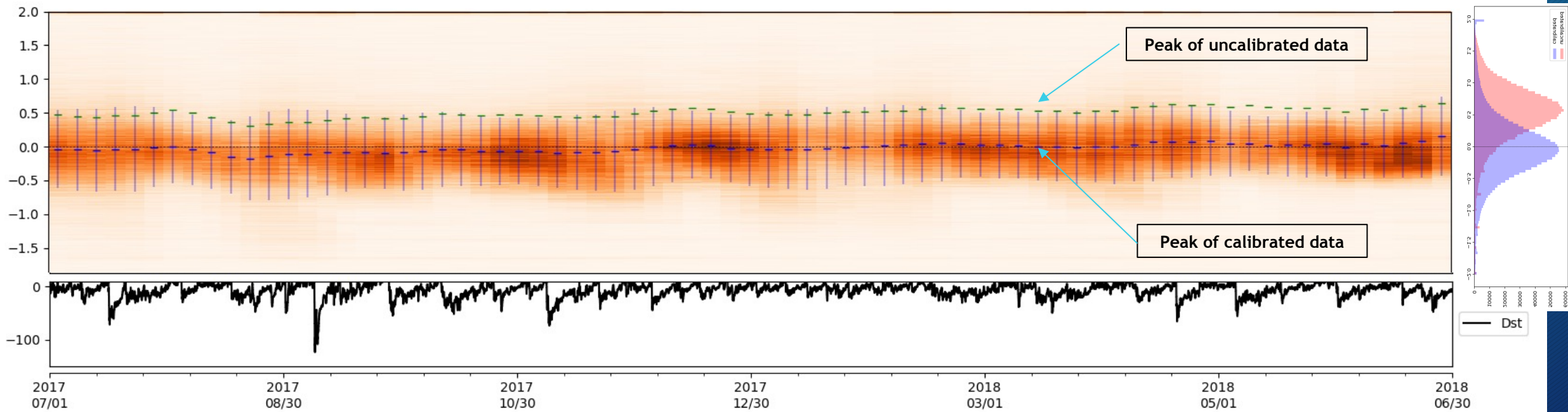


# Calibration results(Long-term, 1year)



$$\text{Comparison ratio} = \frac{d}{r} = \frac{x - y}{x + y}$$

# Calibration results(Long-term, 1year)

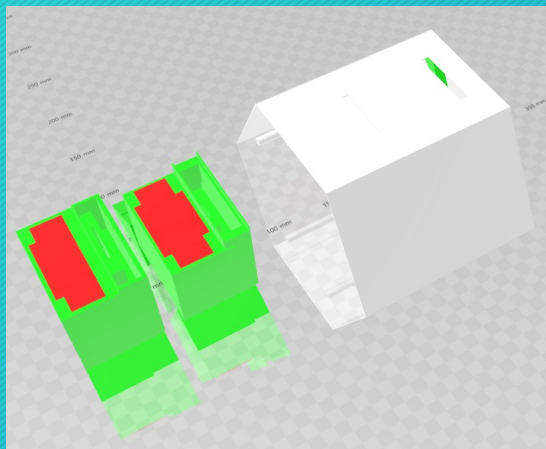


The calibrated data showed a stable state for one year without depending on the Dst index.

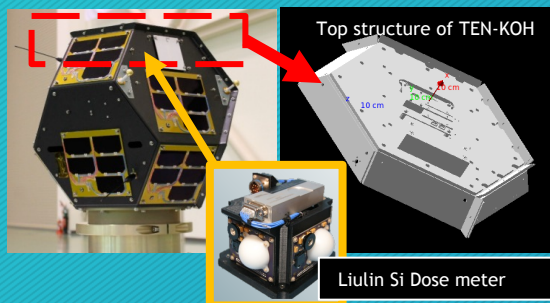
# Advanced Geant4 application development

## Example of GAG simulations

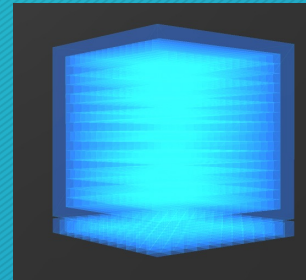
HEP



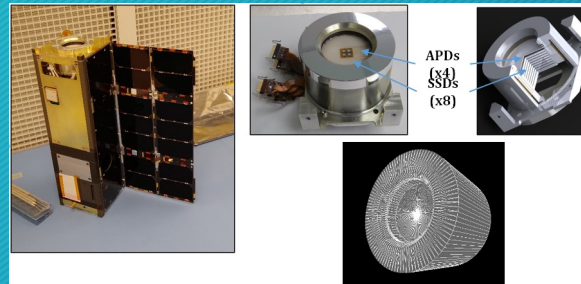
TEN-KOH



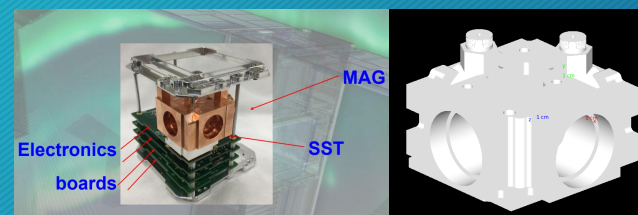
GAGG detector



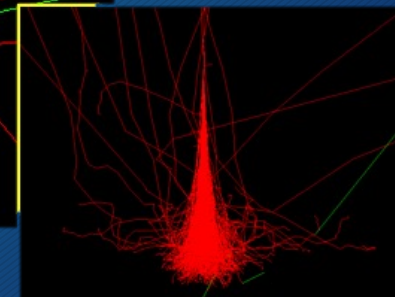
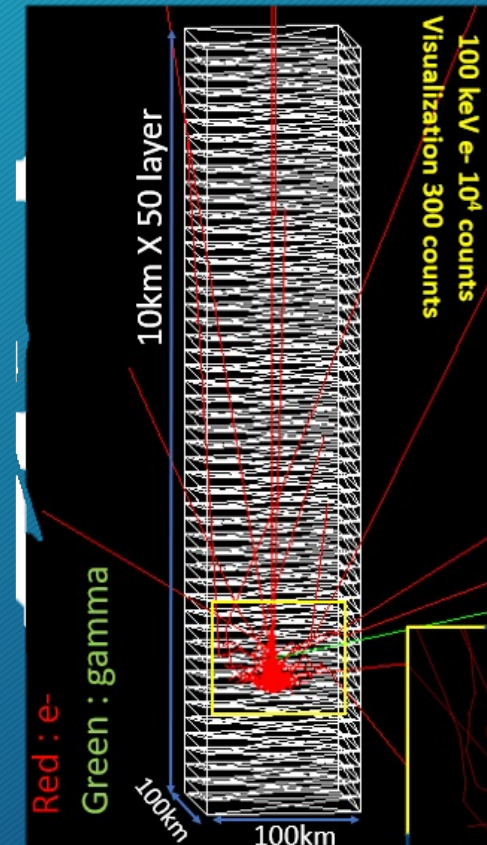
CeREs



SNIPE

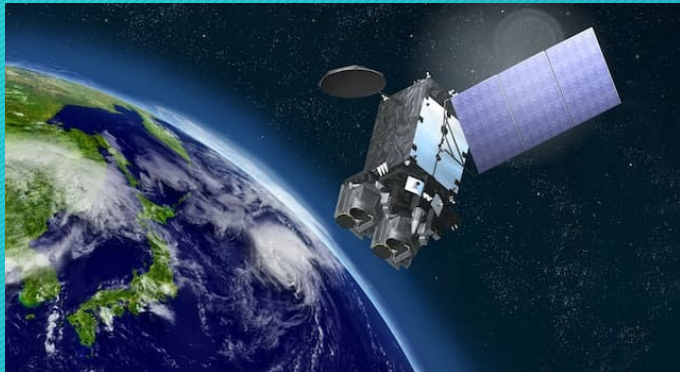


Airshower model



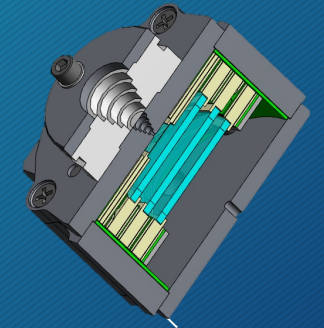
# Calibration of GEO satellite observation

- High-Energy Particle Sensors for Space Weather Applications Onboard the Next Japanese Meteorological Satellite

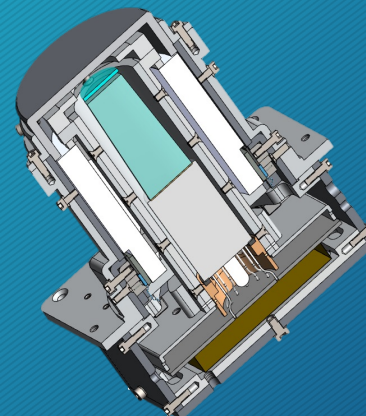


## Radiation Monitors for Space Weather (RMS)

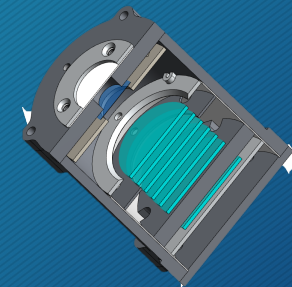
- Operation start : FY2028
- Orbit: GEO
- Observe energy range
  - Proton: 10 MeV ~ 1 GeV
  - Electron: 50 keV~ 5 MeV



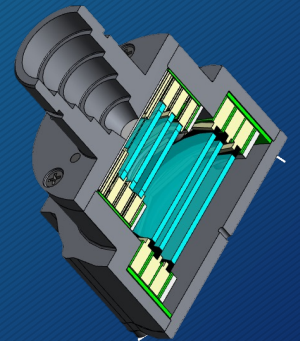
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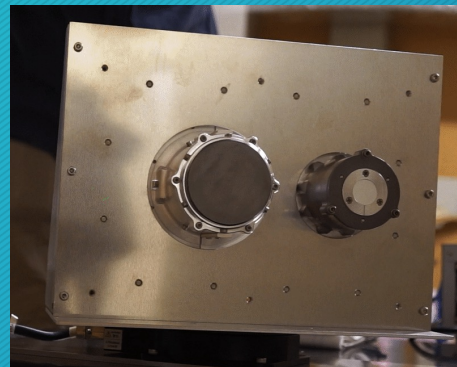
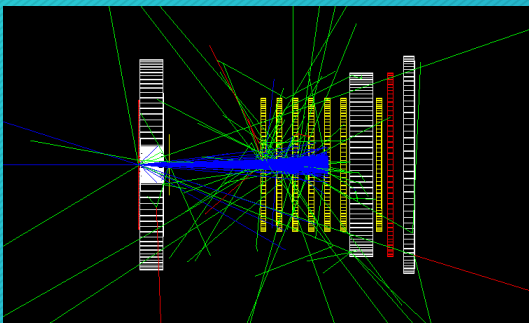
<RMS-p-hi>



<RMS-p-lo>



<RMS-e-hi>



# Summary



- The HEP calibration successfully match to MEP-e observation and it reliable for long term observation, not depending on magnetic storms
- From the inheritance of the calibration, we applied a G4 simulation to several instruments which has complicated geometry.
- The Radiation Monitors for Space Weather(RMS) onboard the next Japanese Meteorological satellite planned to cross calibrate with ARASE or GEO satellite